



Experimental Analysis of Swirl Stabilized Oxyfuel Flames at Elevated Pressure

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Motivation

- Oxyfuel combustion for gas turbines connects the efficient combined cycle process with the post combustion capture option of the Oxyfuel process.
- It is suited for all kinds of fossil fuels, e.g. natural gas or syngas from coal.
- On paper, it is a highly efficient way to avoid CO₂ emissions while using fossil fuels.
- Oxyfuel combustion for gas turbines has never been taken beyond thermodynamic calculations.

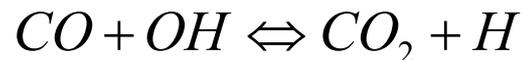
Differences between oxyfuel combustion and air combustion for gas turbines

➤ Differences in physical properties

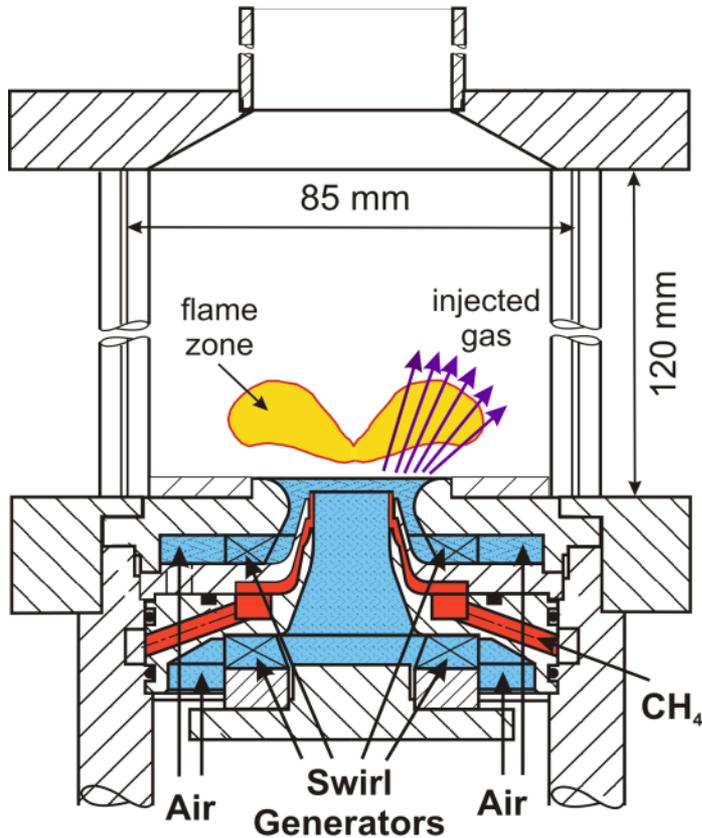
- heat capacity (1.6 times higher than N₂)
- CO₂ is better absorber and emitter of radiative heat
- Gas transport properties (viscosity, diffusivity, thermal conductivity)

➤ Differences in chemical properties

- higher chaperon efficiency of CO₂ (enhances third body reactions)
- Participation in chemical reactions like



Experimental setup



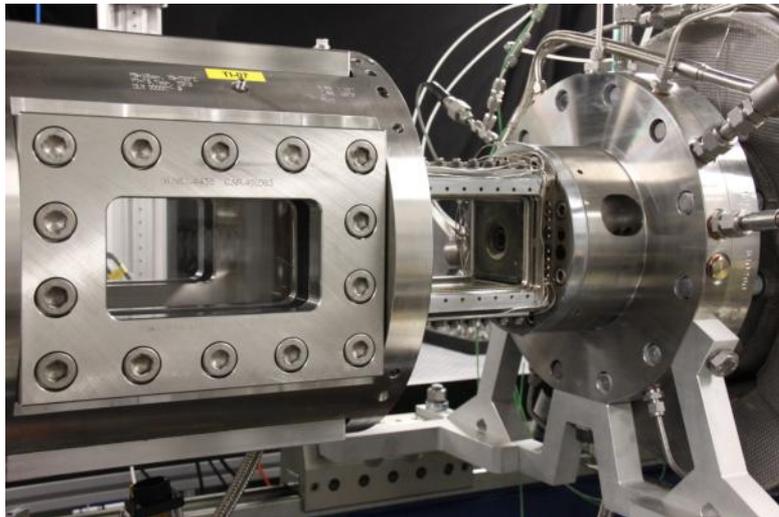
Burner:

- Gas turbine model combustor
- Two co-swirled oxidizer flows
- Fuel injection through a ring between the oxidizer inlets

Operation conditions:

- Fuel: CH₄
- Pressure: 1 - 10 bar
- Thermal power: 10 - 30 kW / bar
- O₂ fraction: 20 % - 40 %
- Fuel / oxidizer ratio: $\phi = 0.5 - 1$
- Fuel / oxidizer temp: 300 - 750 K

High pressure oxyfuel test rig

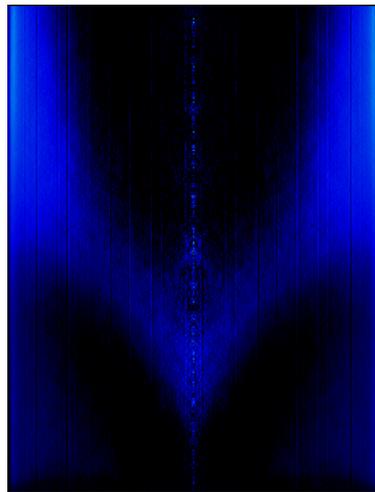


- Pressure: max. 35 bar
- Thermal power: 300 kW

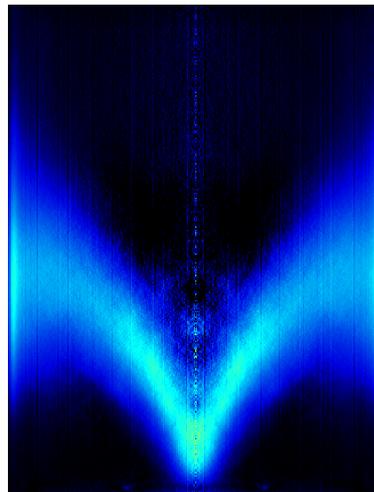
- Oxidizer:
 - Air: 2x 200 g/s
 - Oxygen: 40 g/s
 - CO₂: 120 g/s
 - preheated: up to 800 K

Influence of the oxygen concentration

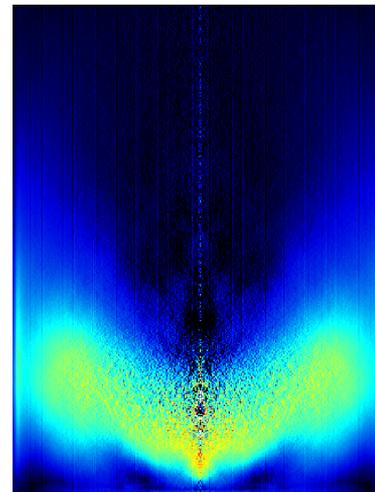
OH*-chemiluminescence imaging, averaged (200 images), Abel transformed



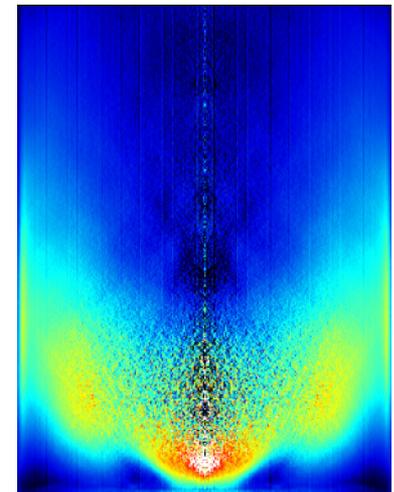
(a) 26 % O₂



(b) 30 % O₂



(c) 34 % O₂

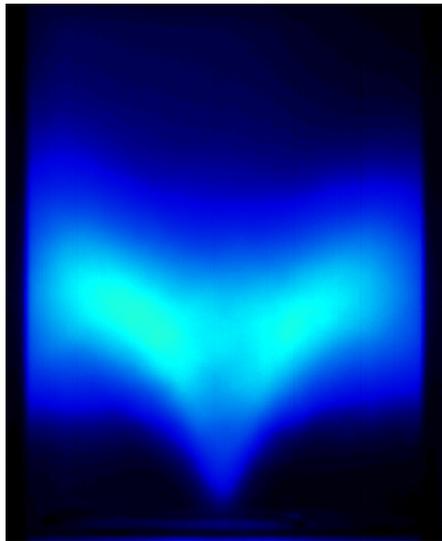


(d) 38 % O₂

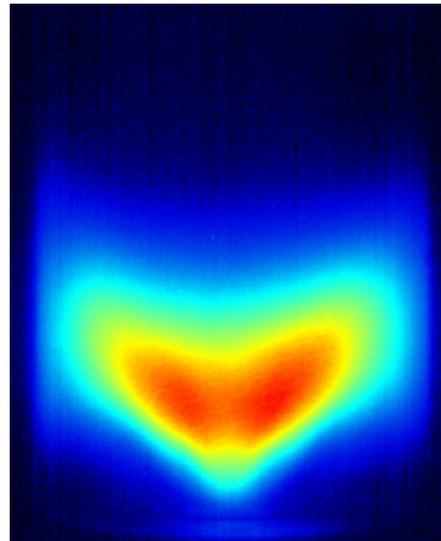
- Oxygen concentration is an additional parameter to influence the combustion behaviour
- Flame speed grows significantly with increasing oxygen level
- To match turbine inlet conditions oxygen concentration is limited to 18 %

Flame stabilization at atmospheric conditions

OH*-chemiluminescence imaging, averaged (200 images)



(a) Ox = 30% O₂
T_{Ox} = 300 K
p = 1 bar
Φ = 0.71
P_{th} = 21.4 kW

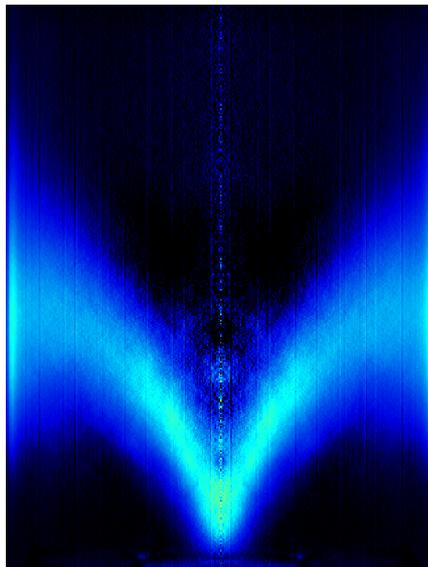


(b) Ox = Air
T_{Ox} = 300 K
p = 1 bar
Φ = 0.70
P_{th} = 20.0 kW

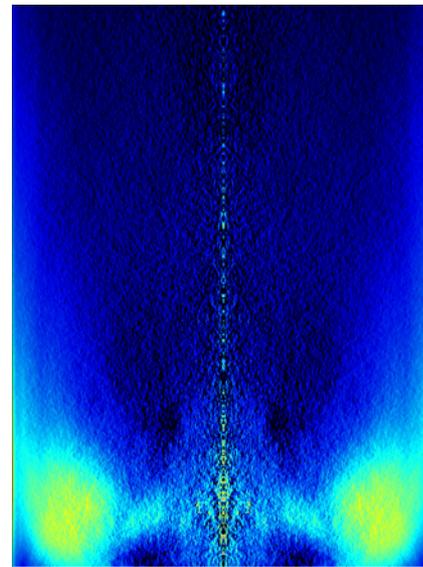
- At 30% O₂ the flame shape resembles the shape of a stable CH₄/ air flame
- The flame stabilizes in the shear layer between irz and inflow of fresh gas

Flame stabilization at pressurized conditions

OH*-chemiluminescence imaging, averaged (200 images), Abel transformed



(a) $O_x = 30\% O_2$
 $T_{Ox} = 300\text{ K}$
 $p = 1\text{ bar}$
 $\Phi = 0.71$
 $P_{th} = 21.4\text{ kW}$

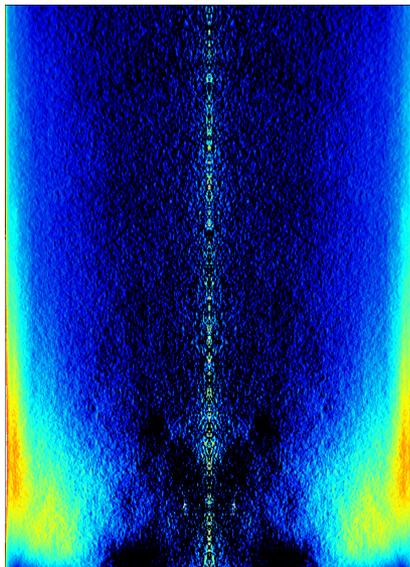


(b) $O_x = 30\% O_2$
 $T_{Ox} = 653\text{ K}$
 $p = 5\text{ bar}$
 $\Phi = 0.71$
 $P_{th} = 157\text{ kW}$

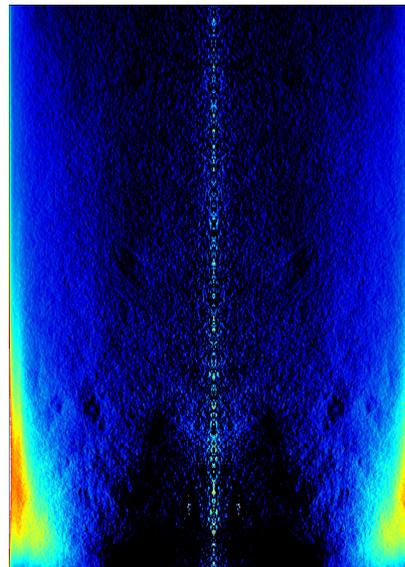
- Flame stabilizes close to the burner surface
- Flame burns mainly in the outer recirculation zone
- No flame stabilization in the shear layer between irz and incoming fresh gas

Flame stabilization at pressurized conditions 2

OH*-chemiluminescence imaging, averaged (200 images), Abel transformed



(a) $O_x = 30\% O_2$
 $T_{Ox} = 653 \text{ K}$
 $p = 5 \text{ bar}$
 $\Phi = 0.95$
 $P_{th} = 157 \text{ kW}$



(b) $O_x = 26\% O_2$
 $T_{Ox} = 753 \text{ K}$
 $p = 5 \text{ bar}$
 $\Phi = 0.90$
 $P_{th} = 157 \text{ kW}$

- At stoichiometric conditions the combustion zone moves closer to the windows
- It was impossible to stabilize flames with less than 26% O_2 in this burner

Strategies for realizing Oxyfuel combustion

➤ Staged combustion

- Flame with higher oxygen concentration and lean conditions
- Fuel injection in the hot products of the first stage

➤ Dilution of the exhaust gas

- Combustion with higher oxygen concentration at stoichiometric conditions
- Dilution of the exhaust gas with CO₂ to reduce temperature

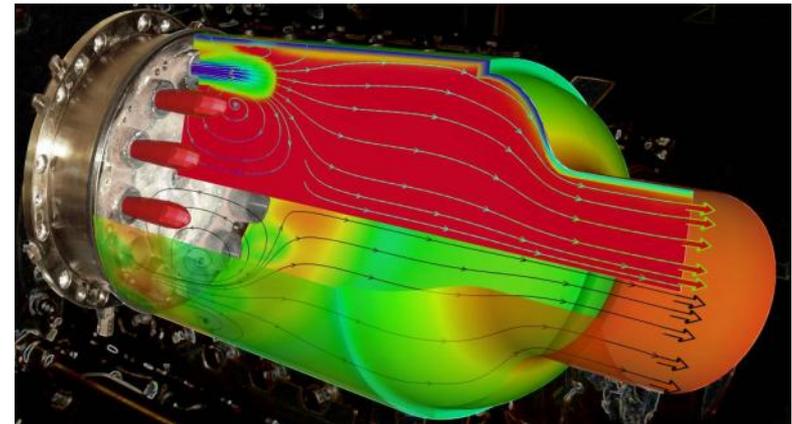
➤ **Alternative combustor design**, which overcomes the problems arising from low flame velocity and reduced ignition delay times



FLOX[®] Burner Concept

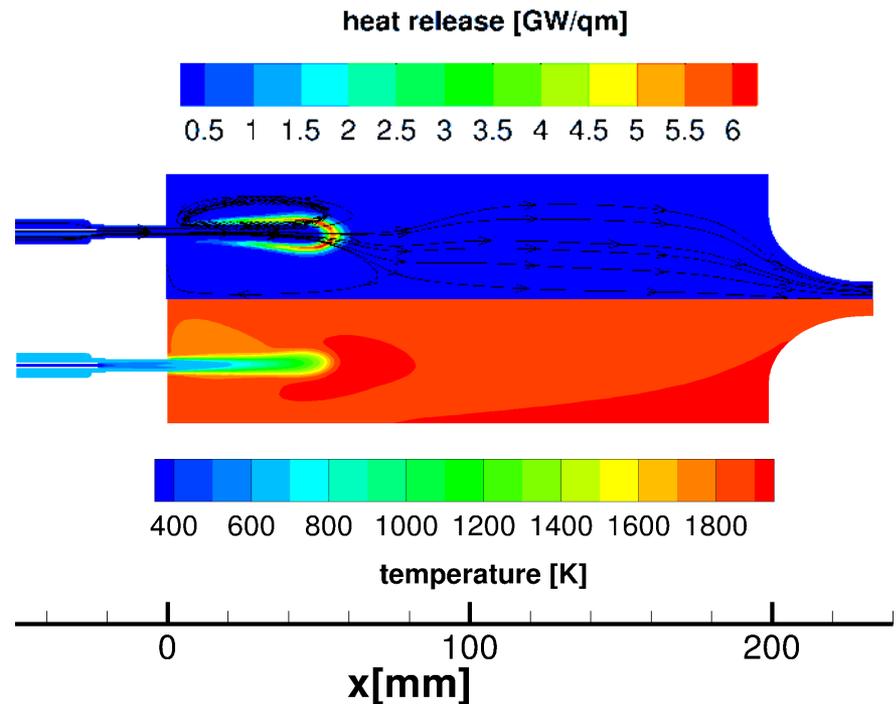
Novel burner concept based on the FLOX[®] principle

- Partially premixed fuel oxidizer mixture
- Discharged through orifices arranged on a circle
- High momentum jets without swirl
- Intense recirculation/mixing in the combustion chamber



FLOX[®] Based Oxyfuel Combustion

- First numerical investigations show:
 - Stable operation with oxygen concentration with 20% O₂ possible
 - Flame stabilization is comparable to methane air flames in the same combustor type



Summary & outlook

- Combustion behaviour of different Oxyfuel / CH₄ flames in a swirl burner was characterized over a wide parameter range at atmospheric and pressurized conditions.
- At pressurized conditions flame stabilizes mainly in the orz.
- It was impossible to stabilize an Oxyfuel flame with 20 % O₂ in the swirl stabilized burner.
- Oxyfuel combustion based on the FLOX® burner concept, could be a promising option
- The new Oxyfuel burner concept has to be tested at pressurized conditions to prove the usability.

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