

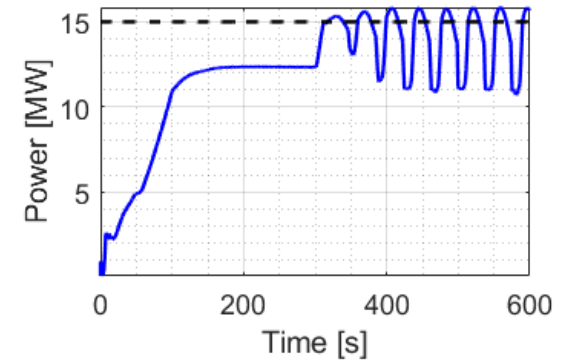
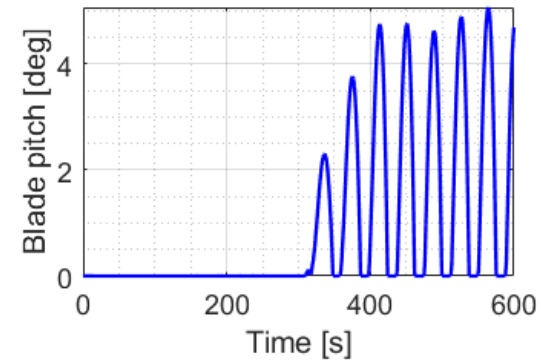
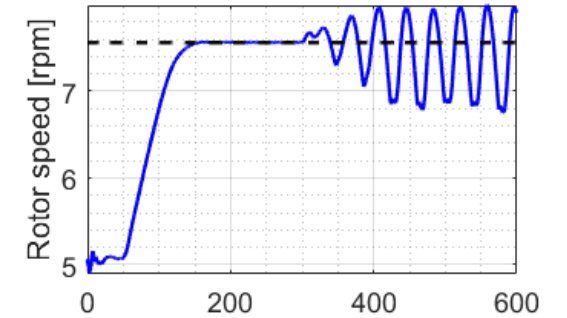
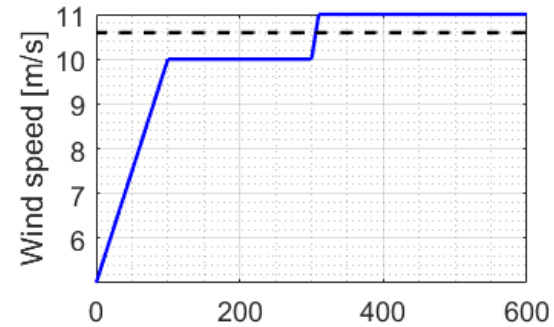
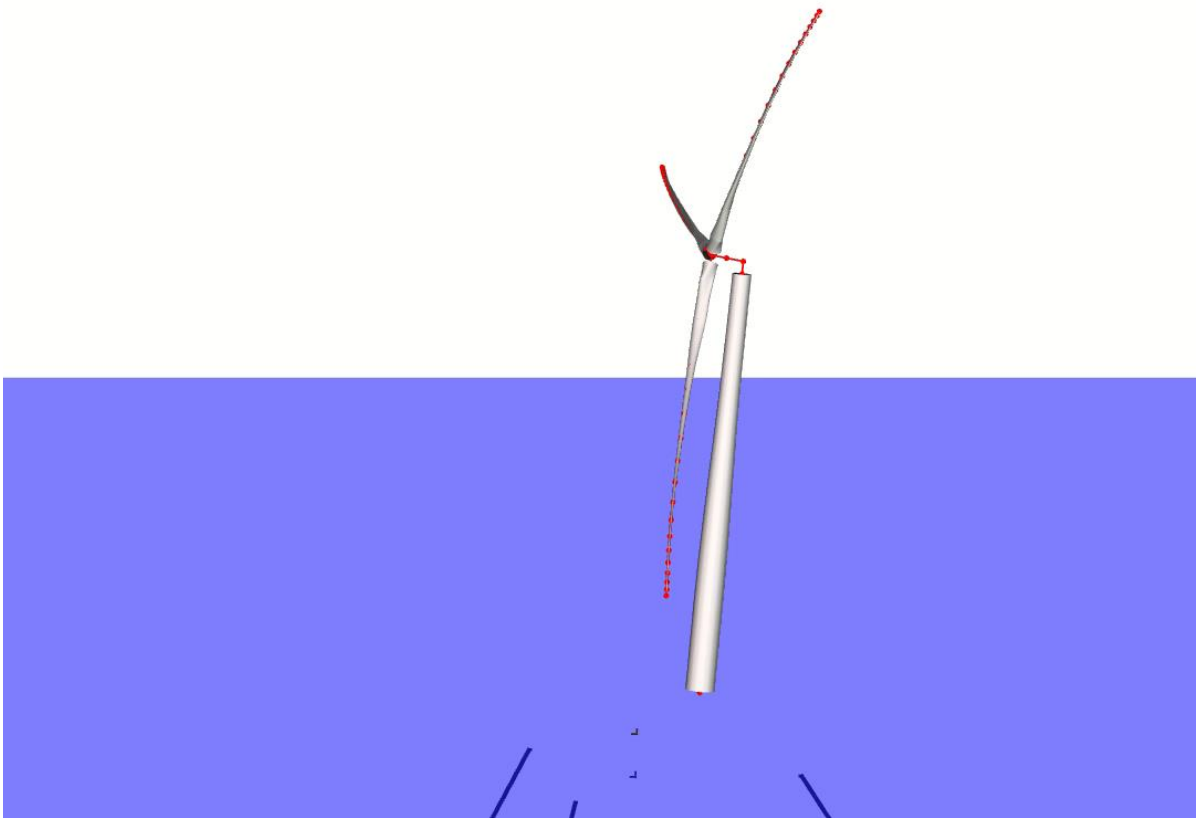
# A control-oriented model for floating wind turbine stability and performance analysis

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DTU Wind and Energy Systems

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# Problem: Floater pitch instability



# Possible solutions

- **De-tuning the controller**
  - We make the controller "slower" than the floater pitch motion
  - Simple solution, but the turbine gets worse at "following the wind"
- **Nacelle-velocity control loops**
  - We include the nacelle velocity in a control feedback loop
    - Loop to blade pitch
    - Loop to generator torque
  - How to find the gains?

Further reading:

- Larsen and Hanson (2007), *A method to avoid negative damped low frequent tower vibrations for a floating, pitch controlled wind turbine*. J. Phys.: Conf. Ser. 75 012073.
- Jonkman (2008), *Influence of control on the pitch damping of a floating wind turbine*. NREL/CP-500-42589, National Renewable Energy Laboratory.
- Yu et al (2018), *Evaluation of control methods for floating offshore wind turbines*. J. Phys.: Conf. Ser. 1104 012033.
- Lenfest et al (2020), *Tuning of nacelle feedback gains for floating wind turbine controllers using a two-dof model*. Proceedings of the ASME 39<sup>th</sup> International Conference on Ocean, Offshore and Arctic Engineering.
- Zhang (2021), *Control design and validation for floating wind turbines*. MSc report M-0472, DTU Wind Energy.

# How to find the gains?

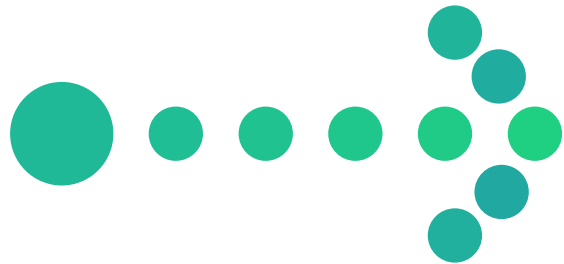
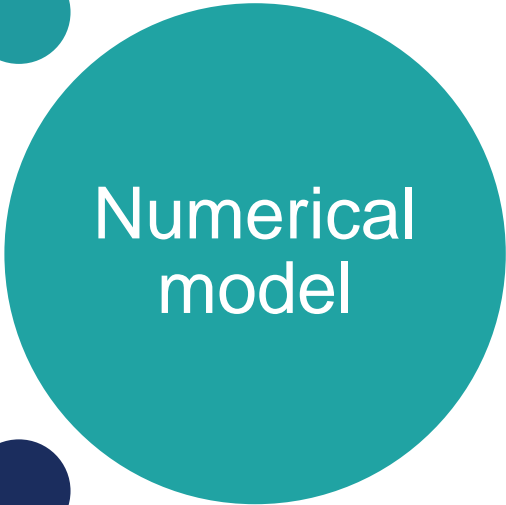
Controller gains



Floating wind turbine



Environmental conditions



Stability

Performance

# How to find the gains?

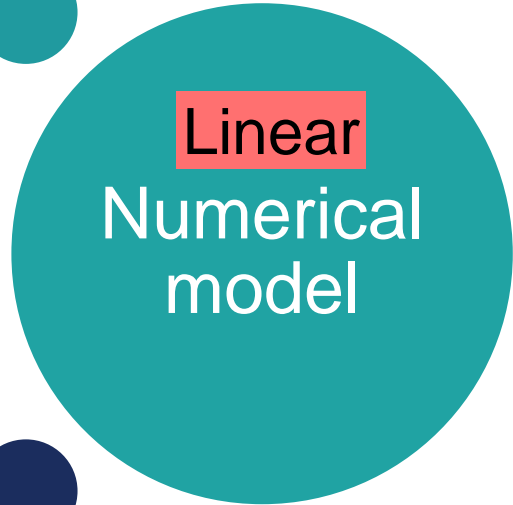
Controller gains



Floating wind turbine

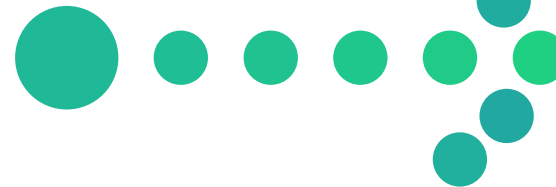


Environmental conditions



Linear

Numerical model



Eigenanalysis



Stability

Performance



Frequency-domain response

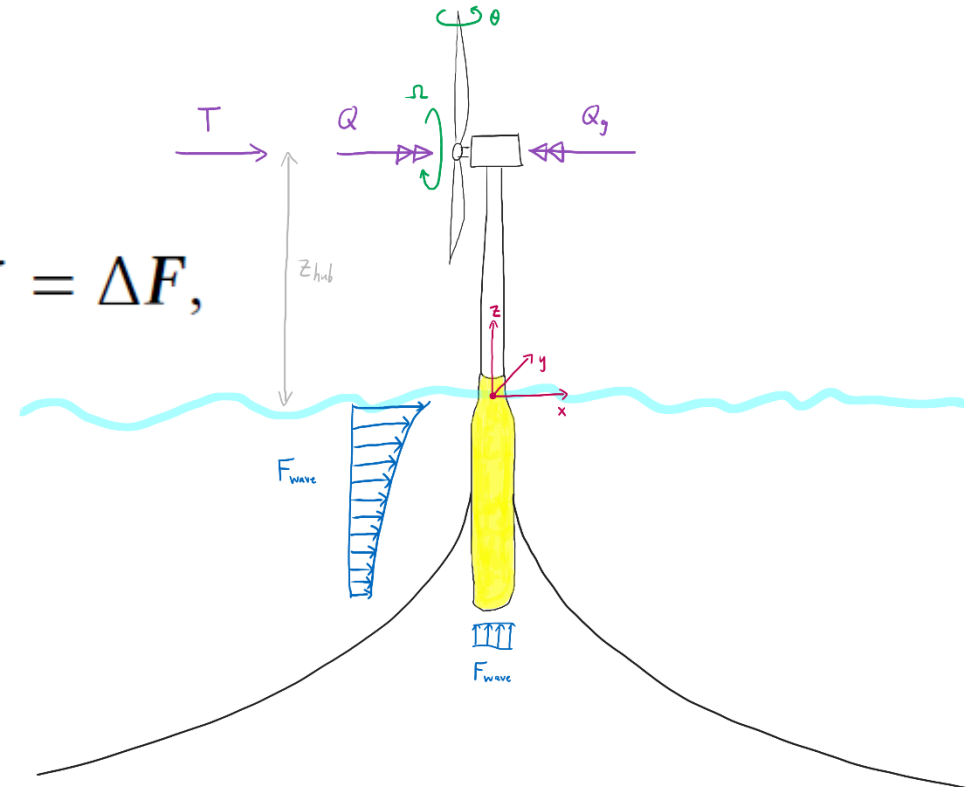
# Floater motion

- Governed by 6x6 system of equations

$$(M + A)\ddot{\xi} + B\dot{\xi} + C\Delta\xi = \Delta F,$$

with:

- Mass matrix
  - Added mass matrix
  - Damping matrix (hydrodynamic)
  - Stiffness matrix (hydrostatic + mooring)
- All variables refer to the deviation from the steady-state



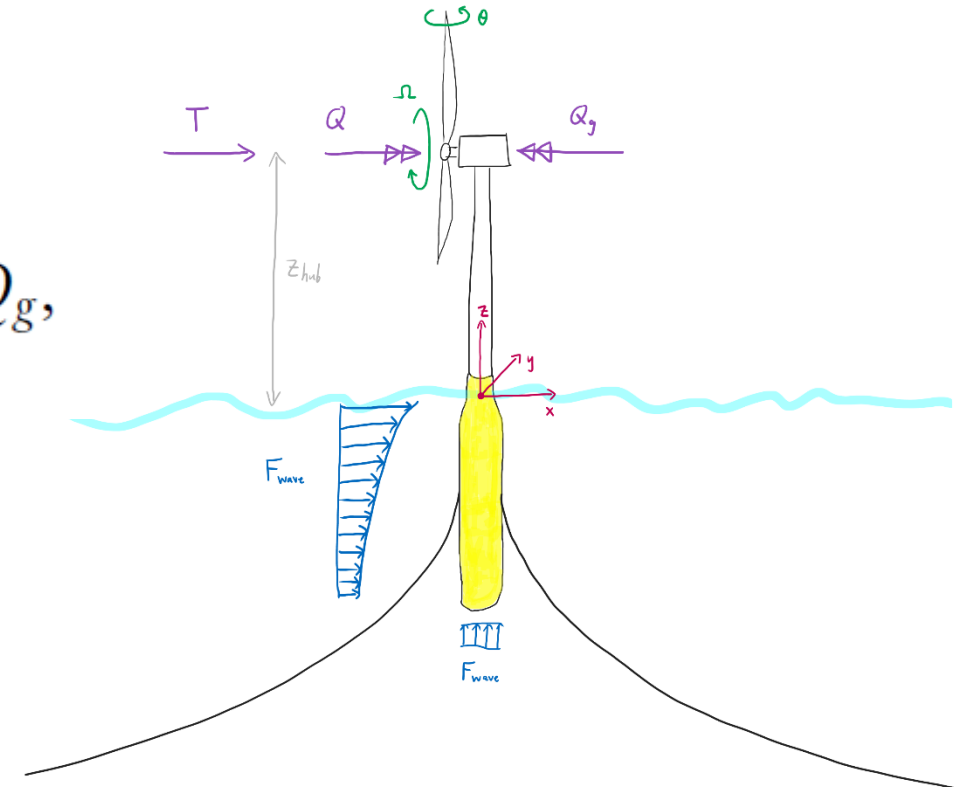
# Drivetrain motion

- Governed by

$$I_{dt} \frac{d\Omega}{dt} = Q - \frac{1}{\eta} Q_g,$$

with:

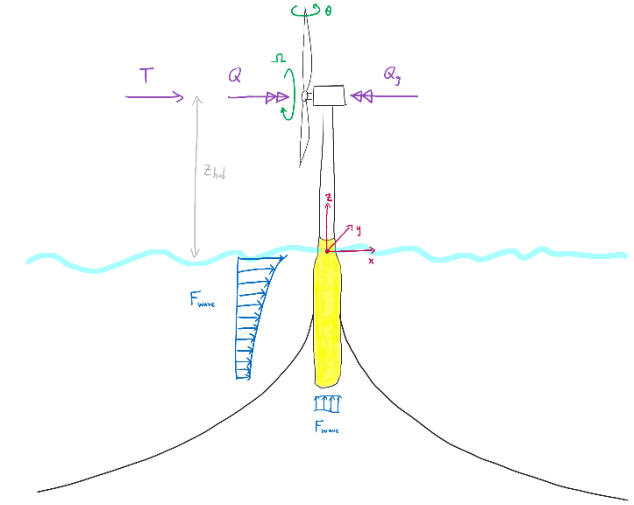
- Drivetrain inertia
- Rotor speed
- Aerodynamic torque
- Generator torque
- Generator efficiency



# Coupling all together

- Fore-aft nacelle velocity

$$\dot{x}_{hub} = \dot{\xi}_1 + z_{hub}\dot{\xi}_5.$$



- Perturbation in wind speed

$$\Delta V = V - V_{op} = V_{turb} - \dot{x}_{hub} = V_{turb} - (\dot{\xi}_1 + z_{hub}\dot{\xi}_5).$$

- Deviation in blade pitch

$$\Delta\theta = \theta - \theta_{op} = k_p\dot{\phi} + k_i\phi + k_b\dot{x}_{hub} = k_p\dot{\phi} + k_i\phi + k_b(\dot{\xi}_1 + z_{hub}\dot{\xi}_5)$$

Error in rotor speed

Proportional Integral

Nacelle-velocity to blade-pitch loop



# Coupling all together

- The aerodynamic torque becomes

The aerodynamic thrust is linearized in a similar way

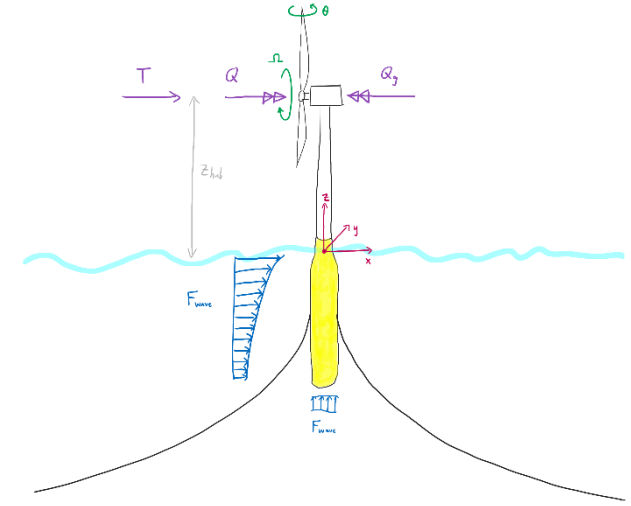
$$Q \approx Q_{op} + \frac{\partial Q}{\partial V} \Delta V + \frac{\partial Q}{\partial \Omega} \dot{\phi} + \frac{\partial Q}{\partial \theta} \Delta \theta.$$

$$Q \approx Q_{op} + \frac{\partial Q}{\partial V} V_{turb} + \left( \frac{\partial Q}{\partial \Omega} + \frac{\partial Q}{\partial \theta} k_p \right) \dot{\phi} + \frac{\partial Q}{\partial \theta} k_i \phi + \left( \frac{\partial Q}{\partial \theta} k_b - \frac{\partial Q}{\partial V} \right) (\dot{\xi}_1 + z_{hub} \dot{\xi}_5).$$

- The generator torque becomes

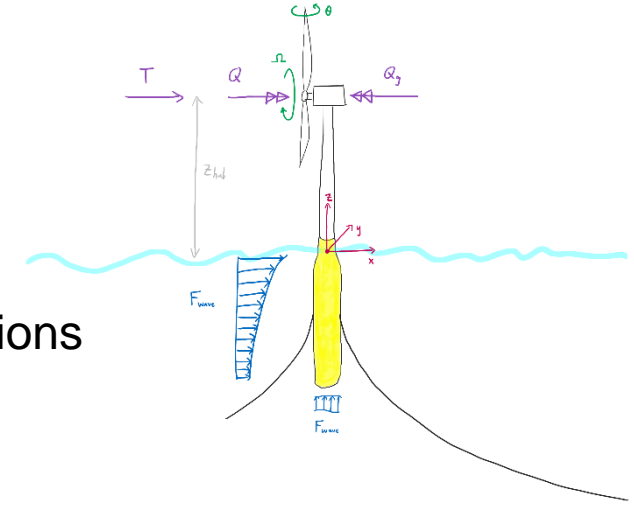
$$Q_g \approx Q_{g,op} + \frac{\partial Q_g}{\partial \Omega} \dot{\phi} - k_q \dot{x}_{hub} = Q_{g,op} + \frac{\partial Q_g}{\partial \Omega} \dot{\phi} - k_q (\dot{\xi}_1 + z_{hub} \dot{\xi}_5)$$

Nacelle-velocity to generator-torque loop



# Coupling all together

- Motion-dependent terms are moved to the left-hand side as contributions to mass, damping or stiffness matrices

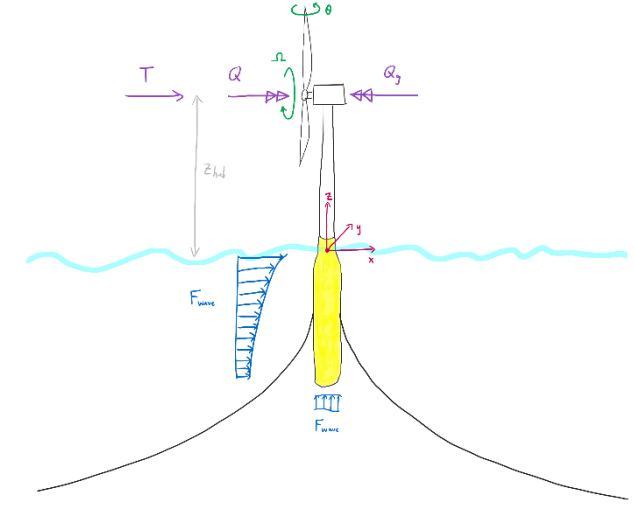


$$(M + A) \begin{bmatrix} \ddot{\xi}_1 \\ \ddot{\xi}_2 \\ \ddot{\xi}_3 \\ \ddot{\xi}_4 \\ \ddot{\xi}_5 \\ \ddot{\xi}_6 \\ \ddot{\phi} \end{bmatrix} + B \begin{bmatrix} \dot{\xi}_1 \\ \dot{\xi}_2 \\ \dot{\xi}_3 \\ \dot{\xi}_4 \\ \dot{\xi}_5 \\ \dot{\xi}_6 \\ \dot{\phi} \end{bmatrix} + C \begin{bmatrix} \Delta \xi_1 \\ \Delta \xi_2 \\ \Delta \xi_3 \\ \Delta \xi_4 \\ \Delta \xi_5 \\ \Delta \xi_6 \\ \phi \end{bmatrix} = \begin{bmatrix} \frac{\partial T}{\partial V} V_{turb} \\ 0 \\ 0 \\ 0 \\ \frac{\partial T}{\partial V} z_{hub} V_{turb} \\ 0 \\ \frac{\partial Q}{\partial V} V_{turb} \end{bmatrix} + F_{wave}.$$

# Coupling all together

- Aero-servo-elastic contributions to mass ( $M$ ), damping ( $B$ ) and stiffness ( $C$ ) matrices

$$C_{ase} = \begin{bmatrix} 0 & 0 & 0 & 0 & 0 & 0 & -\frac{\partial T}{\partial \theta} k_i \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & -\frac{\partial T}{\partial \theta} k_i z_{hub} \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & -\frac{\partial Q}{\partial \theta} k_i \end{bmatrix}$$

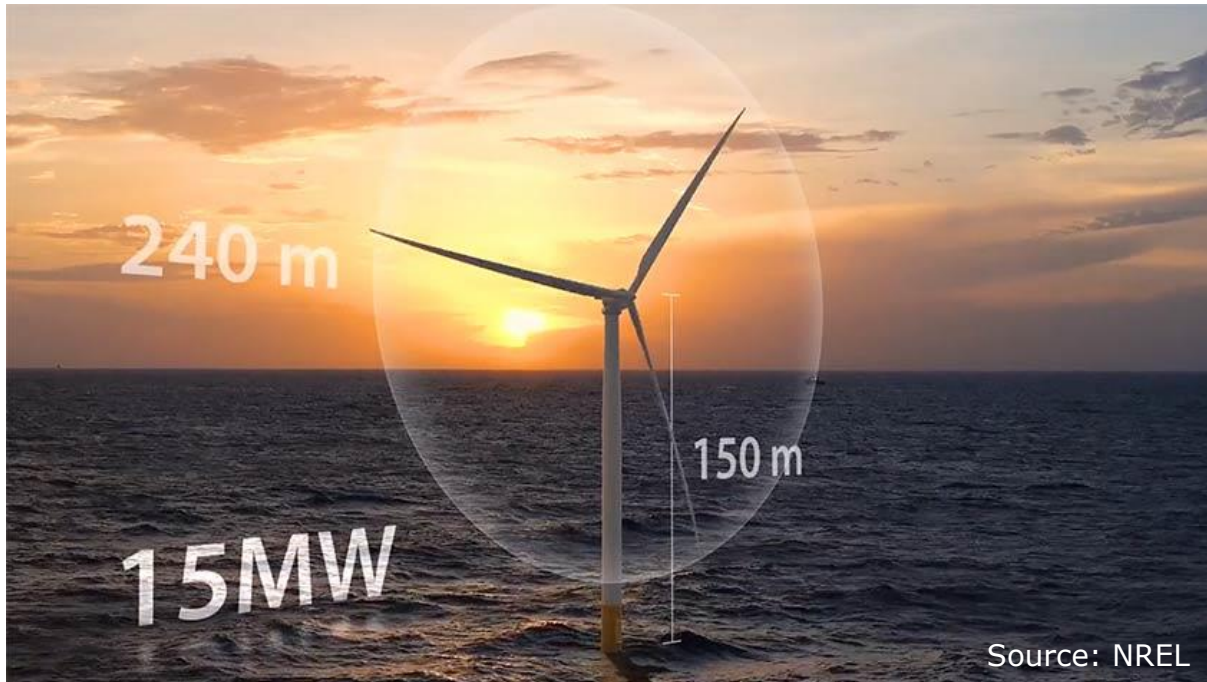


$$M_{ase} = \begin{bmatrix} 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & I_{dt} \end{bmatrix}$$

$$B_{ase} = \begin{bmatrix} \frac{\partial T}{\partial V} - \frac{\partial T}{\partial \theta} k_b & 0 & 0 & 0 & \left(\frac{\partial T}{\partial V} - \frac{\partial T}{\partial \theta} k_b\right) z_{hub} & 0 & -\frac{\partial T}{\partial \Omega} - \frac{\partial T}{\partial \theta} k_p \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ \left(\frac{\partial T}{\partial V} - \frac{\partial T}{\partial \theta} k_b\right) z_{hub} & 0 & 0 & 0 & \left(\frac{\partial T}{\partial V} - \frac{\partial T}{\partial \theta} k_b\right) z_{hub}^2 & 0 & \left(-\frac{\partial T}{\partial \Omega} - \frac{\partial T}{\partial \theta} k_p\right) z_{hub} \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ \frac{\partial Q}{\partial V} - \frac{1}{\eta} k_q - \frac{\partial Q}{\partial \theta} k_b & 0 & 0 & 0 & \left(\frac{\partial Q}{\partial V} - \frac{1}{\eta} k_q - \frac{\partial Q}{\partial \theta} k_b\right) z_{hub} & 0 & \frac{1}{\eta} \frac{\partial Q_g}{\partial \Omega} - \frac{\partial Q}{\partial \Omega} - \frac{\partial Q}{\partial \theta} k_p \end{bmatrix}$$

# Our case study

- IEA Wind 15 MW turbine



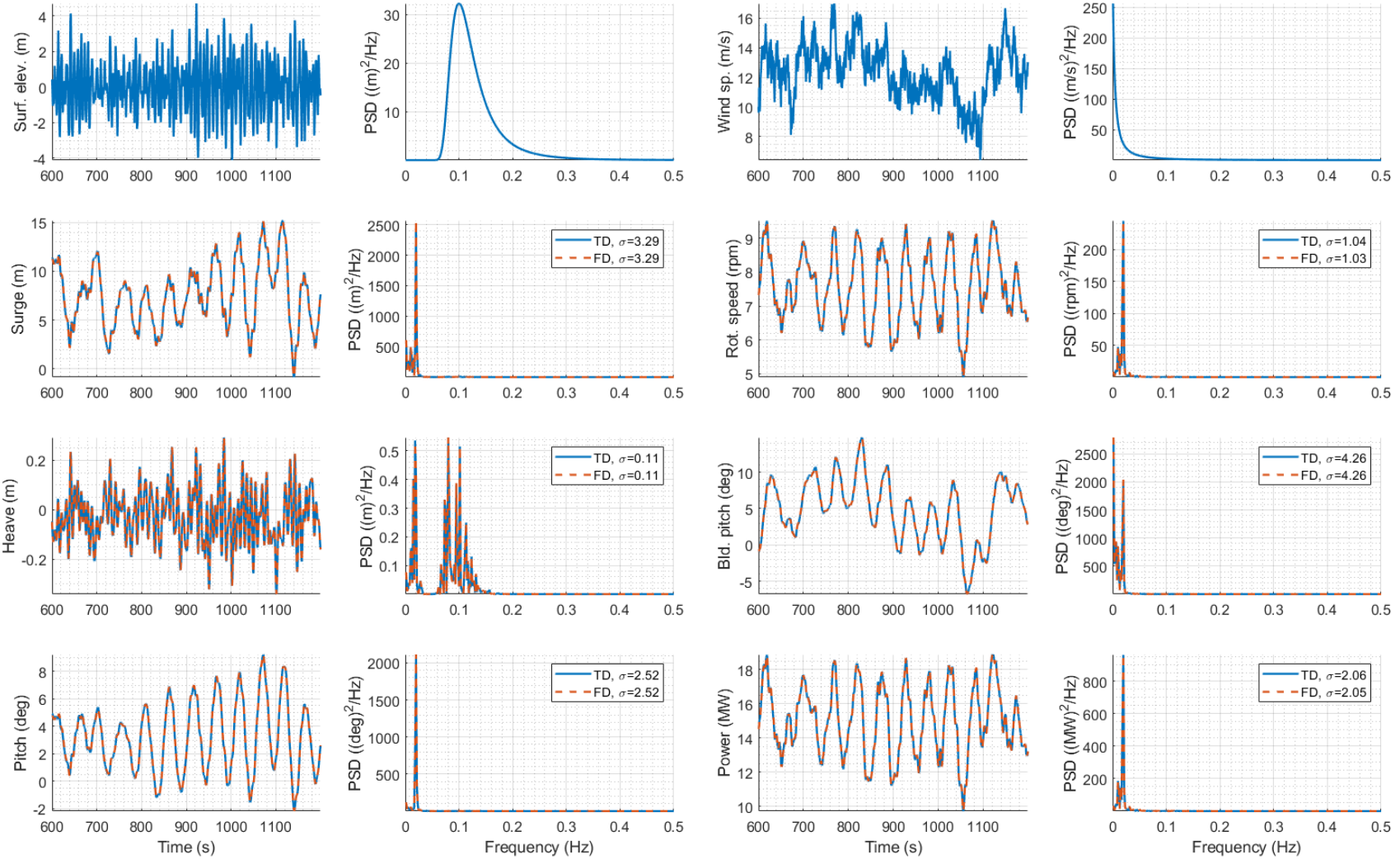
- WindCrete spar-buoy floater



# Response to wind and waves

De-tuned controller

Controller with  $f_c=0.02$  Hz,  $\zeta=0.7$ ,  $k_b=0.00$  rad/(m/s),  $k_q=0e+00$  Nm/(m/s)

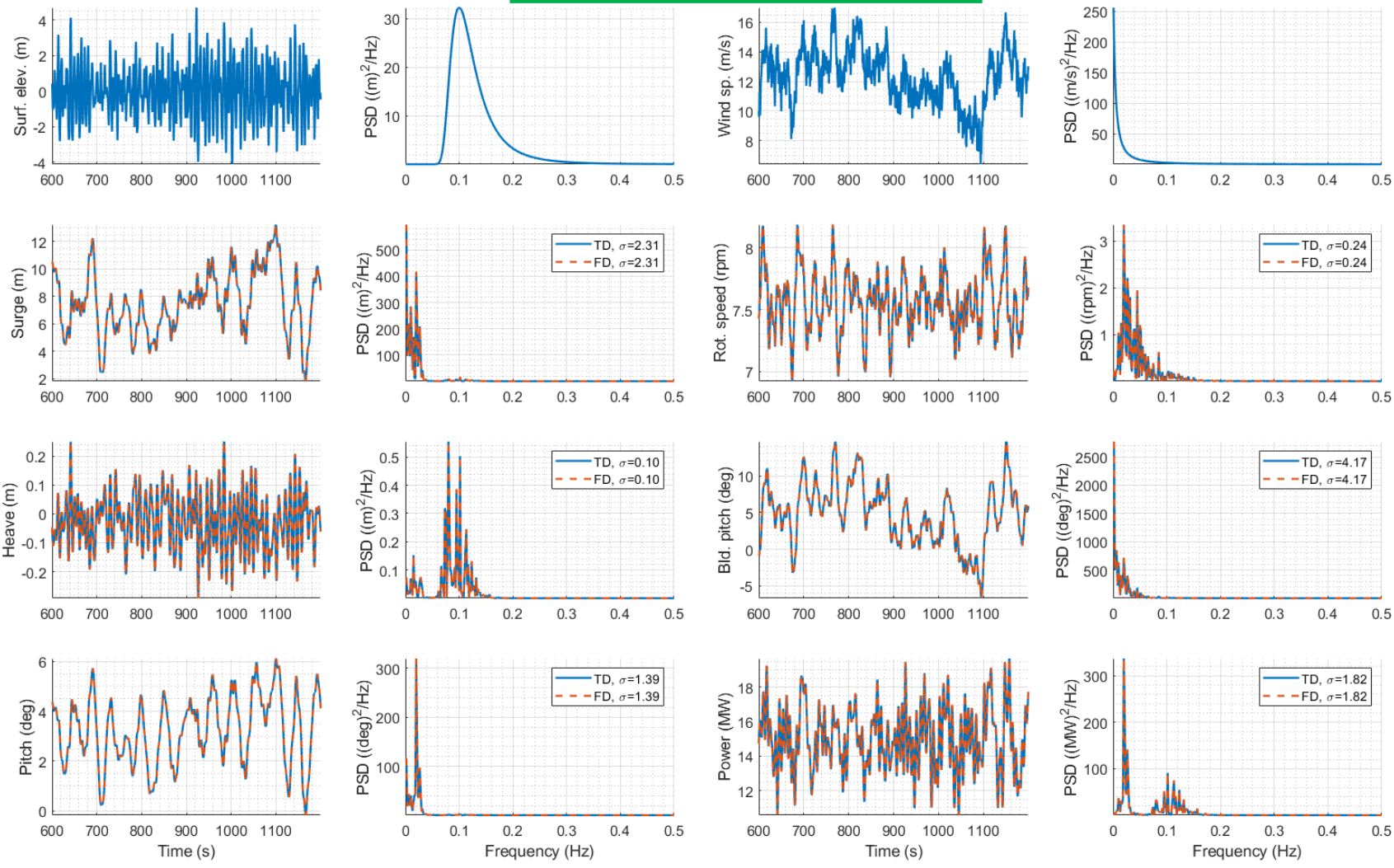


Pitch damping ratio 3.75%

# Response to wind and waves

Nacelle-velocity feedback to generator torque active

Controller with  $f_c=0.05$  Hz,  $\zeta=0.7$ ,  $k_b=0.00$  rad/(m/s),  $k_q=2e+06$  Nm/(m/s)

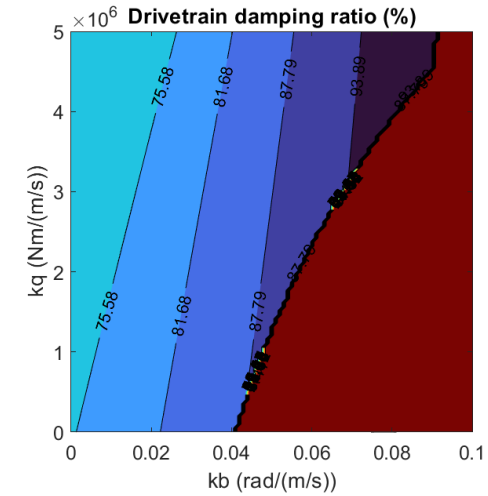
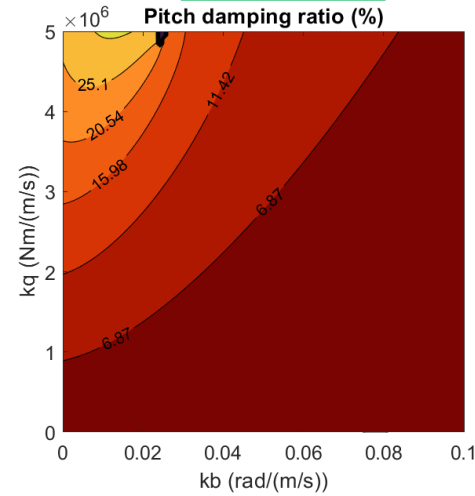
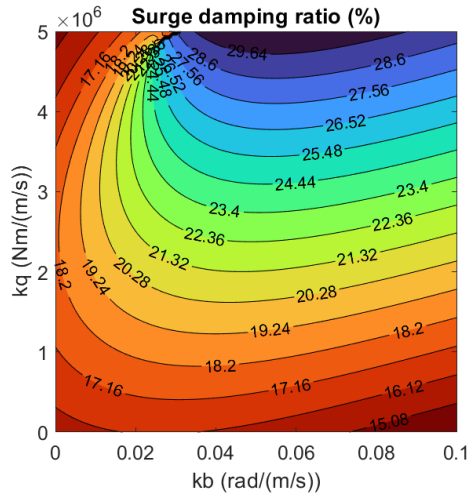


Pitch damping ratio 10.25%

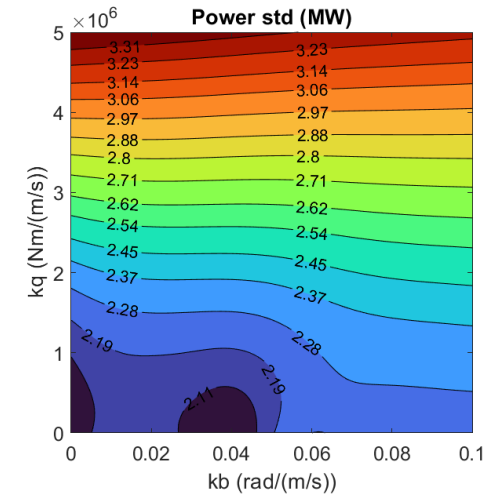
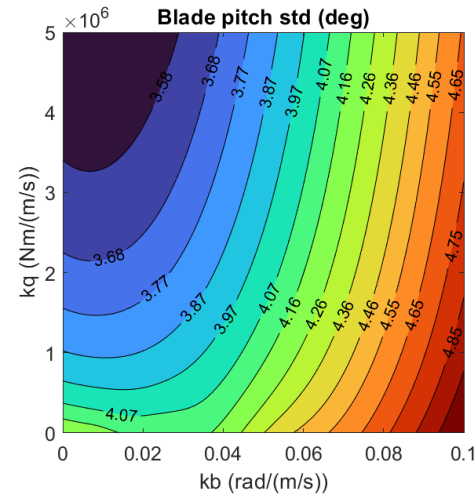
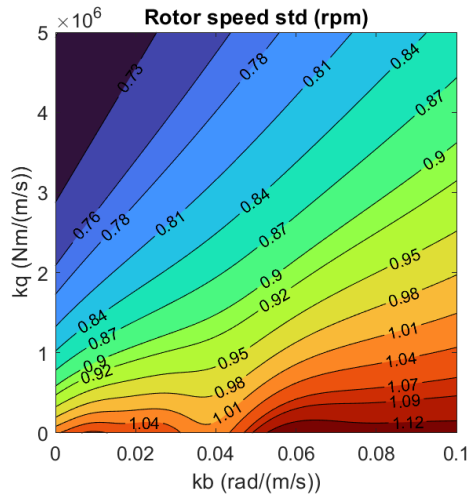
# Parametric study on 100x100 grid

Onshore gains set for  $f_c=0.02$  Hz,  $\zeta=0.7$   $V=12$  m/s

Damping ratios  
(blue=larger)



Standard deviations  
(blue=smaller)

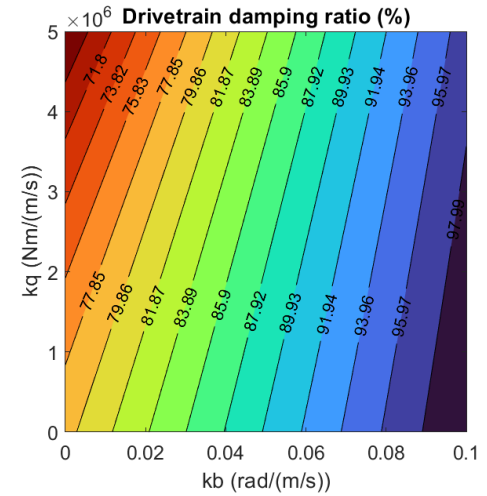
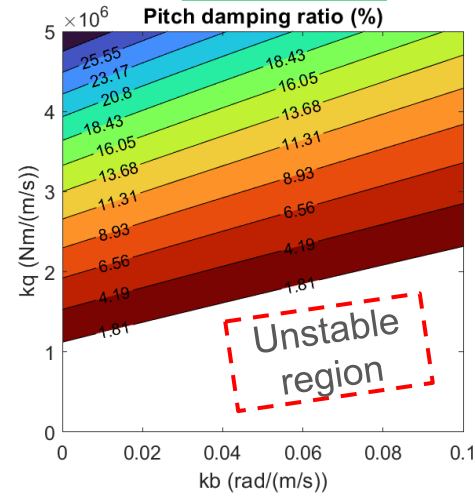
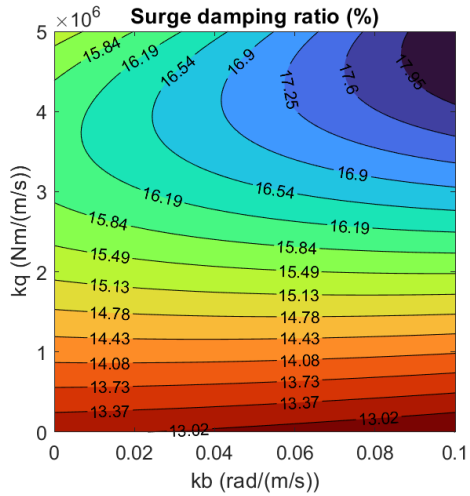


10.000  
controllers  
in ~3 min

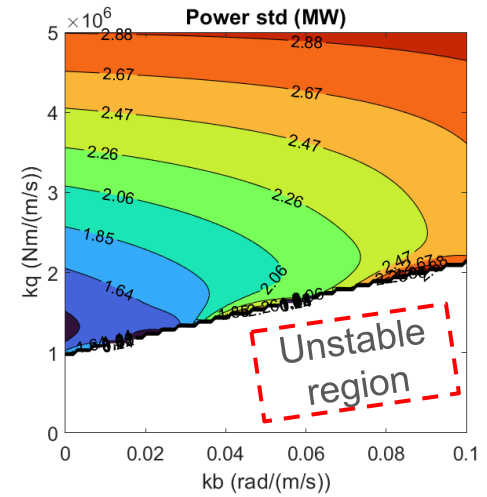
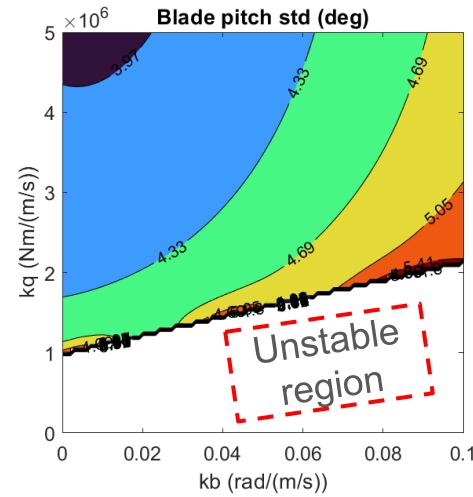
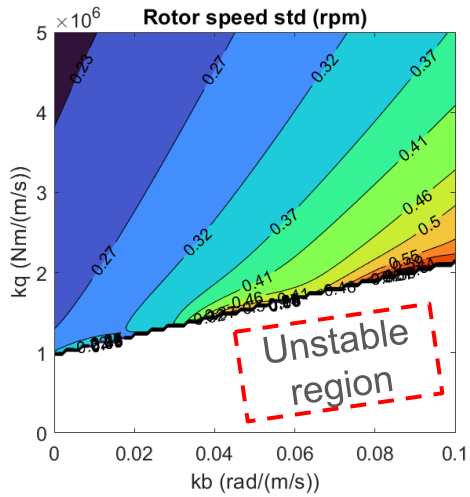
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Onshore gains set for  $f_c=0.05$  Hz,  $\zeta=0.7$   $V=12$  m/s

Damping ratios  
(blue=larger)



Standard deviations  
(blue=smaller)



10.000  
controllers  
in ~3 min



# Conclusions

- Model to investigate controller stability and performance for floating wind turbines
- Classic PI gains + nacelle-velocity feedback gains
  - Blade pitch
  - Generator torque
- Efficient evaluation thanks to linear response formulation
  - 10.000 controllers in ~3 min
- Must be followed by analysis in time-domain model, including:
  - Realistic wind loads
  - Actuator dynamics
  - Nonlinear effects
  - etc

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