



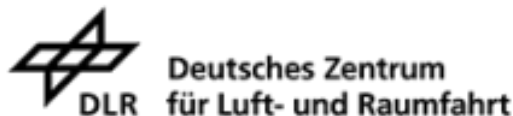
A new framework for aeroelastic simulation of offshore wind turbine megastructures

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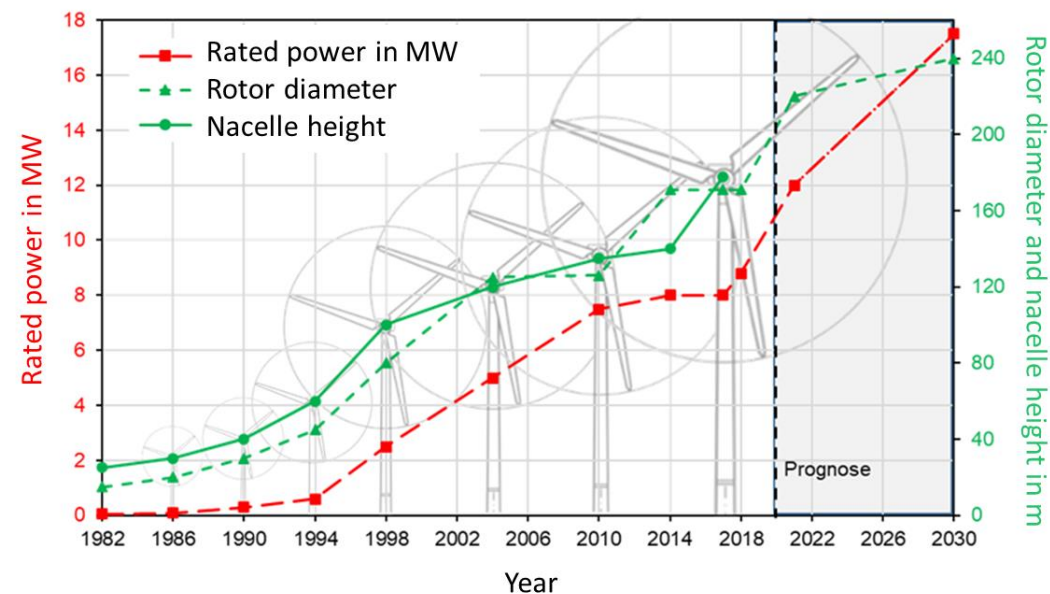
Founded by the Deutsche Forschungsgemeinschaft (DFG, German Research Foundation), Project-ID 434502799 – SFB 1463





Motivation

- Increasing structure sizes of future wind turbines due to the need for efficiently produced clean energy
- Instead of upscaling new designing methods are needed
- Developing a simulation tool to analyze the aeroelastic behavior including all nonlinear dynamic interactions, e.g.
 - Structural nonlinearities
 - Fluid-Structure-Interaction
 - Wind turbine control
- Simulation framework DeSiO
 - State of the art mid-fidelity methods
 - Computationally efficient yet sufficiently accurate



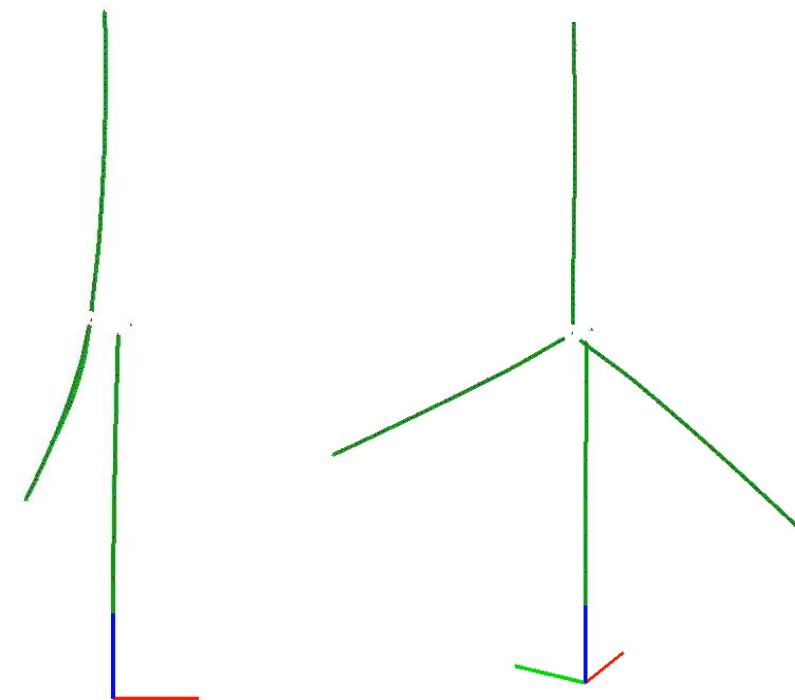


Structural Model

- Multibody system for rigid and flexible structures
- Governing equation derived using Hamilton's principle
- Total Lagrangian Description
- Director-based kinematics
- Holonomic and non-holonomic constraints
- Energy-conserving/dissipative, momentum-conserving time integration method
- Several analyses: nonlinear static, nonlinear dynamic, buckling and modal analyses

$$\mathcal{S} = \int_{t_1}^{t_2} \left(\underbrace{\mathcal{K}(\mathbf{v}; t)}_{\text{kinetic energy}} - \underbrace{[\mathcal{W}(\mathbf{x}; t) - \mathcal{P}(\mathbf{x}; t)]}_{\text{external work}} + \underbrace{\boldsymbol{\lambda}(t) \cdot \mathbf{h}(\mathbf{x}; t)}_{\text{constraints}} + \mathbf{v}(t) \cdot \underbrace{[\mathbf{l}(\mathbf{v}; t) - \mathbf{l}(\dot{\mathbf{x}}; t)]}_{\text{momentum}} \right) dt$$

$$\delta \mathcal{S} = 0$$



Left: Modal analysis, first eigenmode
Right: Transient simulation
NREL 15 MW turbine





Aerodynamic Model

- We use the Unsteady Vortex Lattice Method (UVLM)
- Computationally fast yet sufficiently accurate
- Advantages of this method are e.g., (compared with the Blade Element momentum theory):
 - More accurate realization of aerodynamic response
 - Full free wake, unsteady effects
 - Considering aeroelastic stability effects
 - No restriction of camber and/or angle of attack
- Assumptions
 - Incompressibility of the fluid
 - Neglecting viscous phenomena like turbulences

Non penetration condition: $\mathbf{v} \cdot \mathbf{n} = 0$

$$\mathbf{v} = \underbrace{\mathbf{v}_\infty}_{\text{free field}} + \underbrace{\mathbf{v}_s}_{\text{surface}} + \underbrace{\mathbf{v}_{v,surf}}_{\text{vortex-induced surface}} + \underbrace{\mathbf{v}_{v,wake}}_{\text{vortex-induced wake}}$$

$$\mathbf{v}_v(\mathbf{r}; t) = \frac{\Gamma}{4\pi} \frac{(\mathbf{r}_1 + \mathbf{r}_2) \mathbf{r}_1 \times \mathbf{r}_2}{r_1 r_2 (r_1 r_2 + \mathbf{r}_1 \cdot \mathbf{r}_2) + r_c^2}$$





Fluid-Structure-Interaction

- A weak coupling between the **structural** and **aerodynamic** models leads to a slow convergence towards the exact solution
 - Weak coupling: Structural and aerodynamic contributions are calculated at different time steps: f_A^n, f_S^{n+1}
- A strong coupling between the **structural** and **aerodynamic** models leads to a faster convergence towards the exact solution
 - Strong coupling: Structural and aerodynamic contributions are calculated at the same time step: f_A^{n+1}, f_S^{n+1}
- The nonlinear governing equation is solved using Newton's method: Therefore the aerodynamic load needs to be linearized along the strong coupling

<p>Schematic structure of a weak coupling</p>	<p>Governing equation for the weak coupling</p> $g_{n+\frac{1}{2}}^{weak} = \begin{bmatrix} f_d^{int}(q_{n+1}, q_n, s_{n+1}, s_n) - f_d^{ext}(q_{n+1}, q_n) + \dot{l}_d(s_{n+1}, s_n) + H_d^T(q_{n+1}, q_n) \cdot \lambda_{n+1} - f_A(q_n, s_n) \\ h_d(q_{n+1}, q_n) \\ l_d(s_{n+1}, s_n) - l_d(q_{n+1}, q_n) \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix}$
<p>Schematic structure of a strong coupling</p>	<p>Governing equation for the strong coupling</p> $g_{n+\frac{1}{2}}^{strong} = \begin{bmatrix} f_d^{int}(q_{n+1}, q_n, s_{n+1}, s_n) - f_d^{ext}(q_{n+1}, q_n) + \dot{l}_d(s_{n+1}, s_n) + H_d^T(q_{n+1}, q_n) \cdot \lambda_{n+1} - f_A(q_{n+1}, q_n, s_{n+1}, s_n) \\ h_d(q_{n+1}, q_n) \\ l_d(s_{n+1}, s_n) - l_d(q_{n+1}, q_n) \end{bmatrix} = \begin{bmatrix} 0 \\ 0 \\ 0 \end{bmatrix}$