

Real-time hybrid model testing of floating wind turbines: sensitivity to limited actuation

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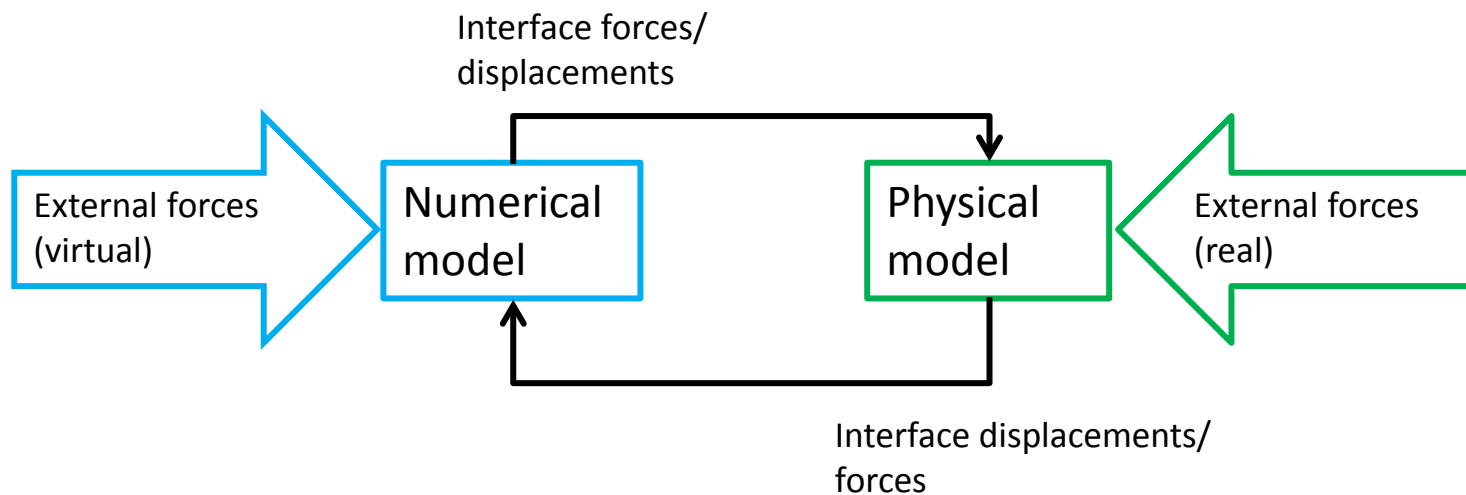
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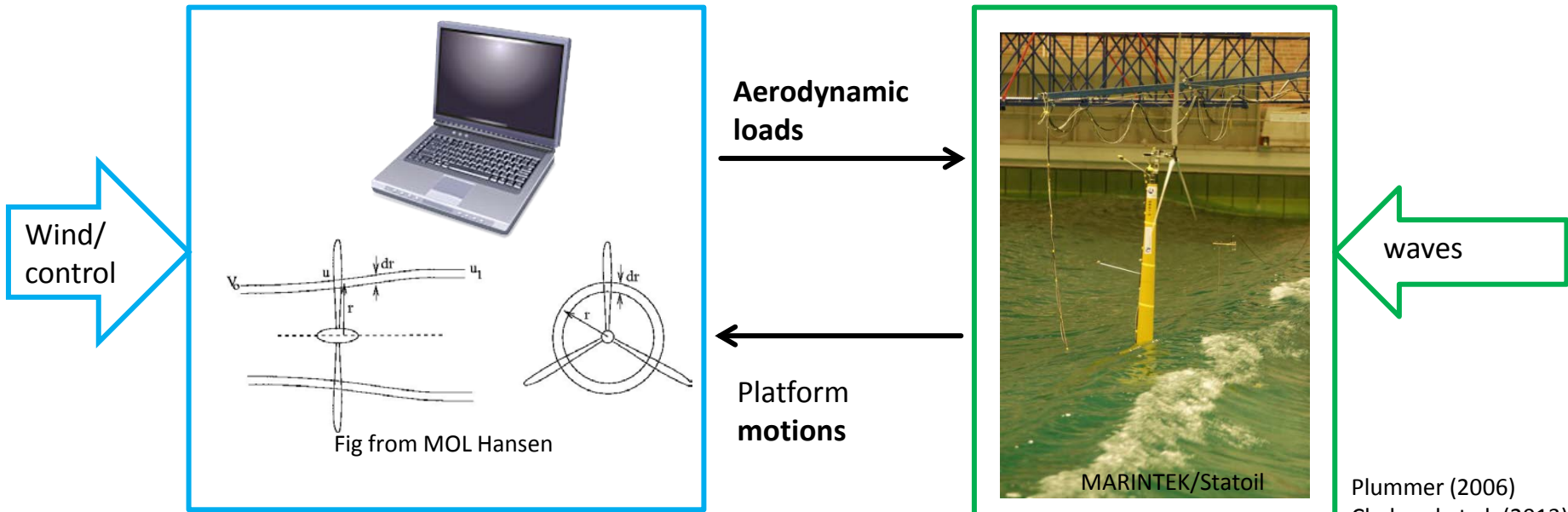
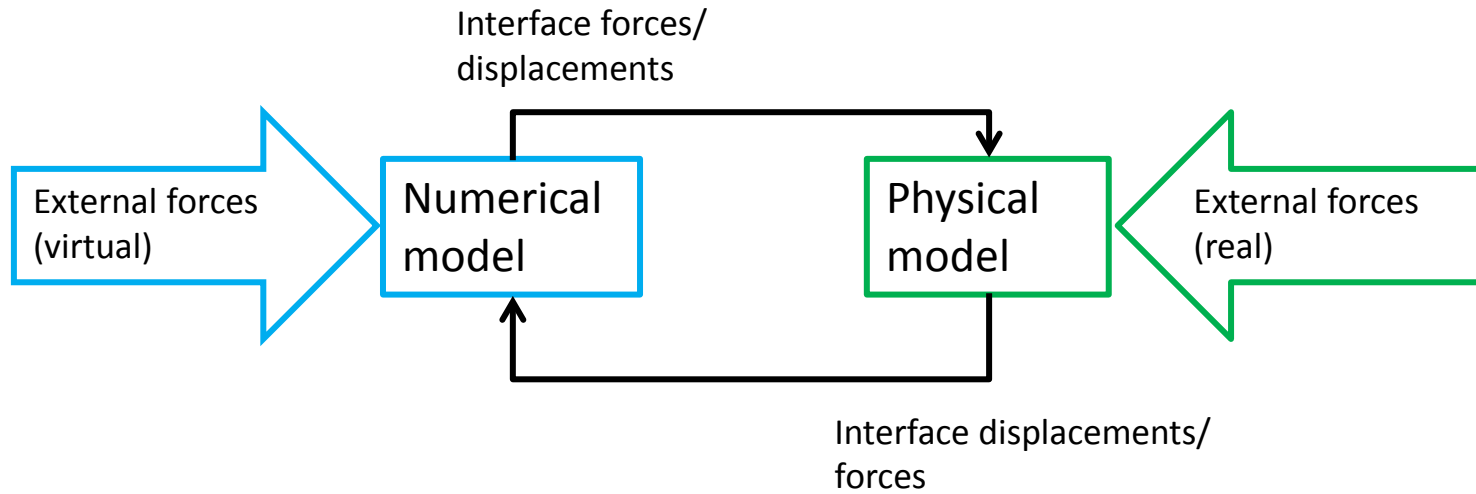
 **SINTEF**

Real-time hybrid model testing (ReaTHM)

- Also called Hardware-in-the-Loop (HiL) or Model-in-the-Loop (MiL)
- The test specimen is part real and part virtual
- Only the key component with unknown dynamics is physically tested
- Some conditions may be more accurately represented by a computer model than by laboratory conditions
- Characteristics of the simulated system can be easily varied



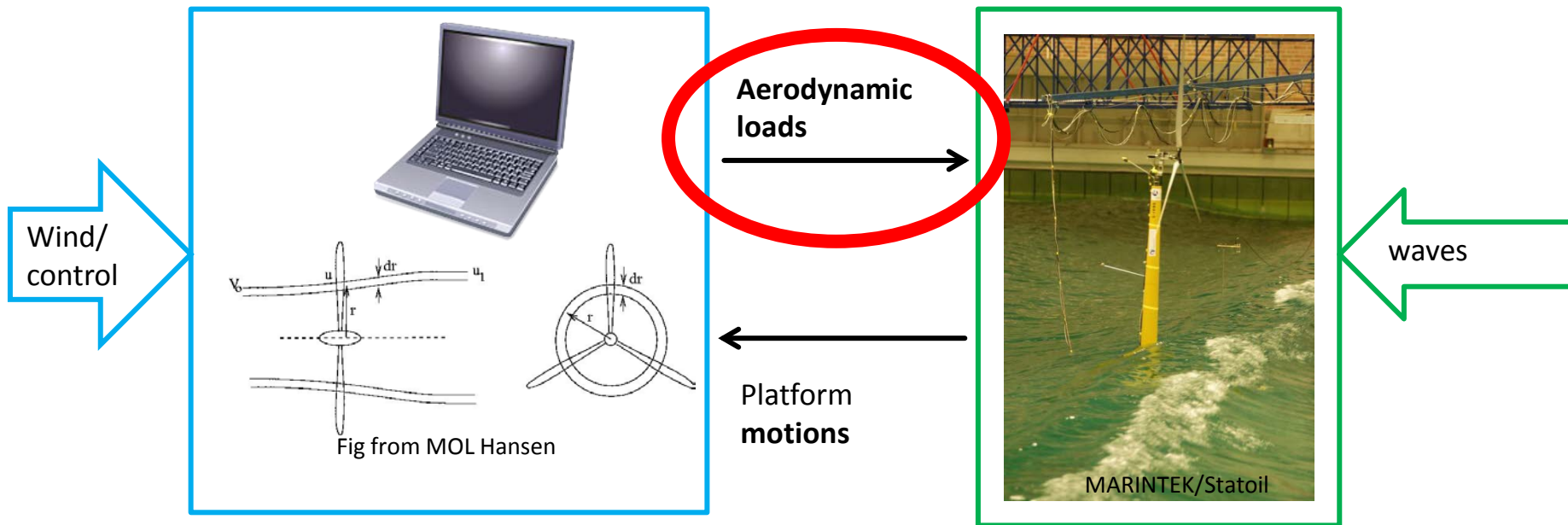
Plummer (2006)
Chabaud et al. (2013)
Hall et al. (2014)



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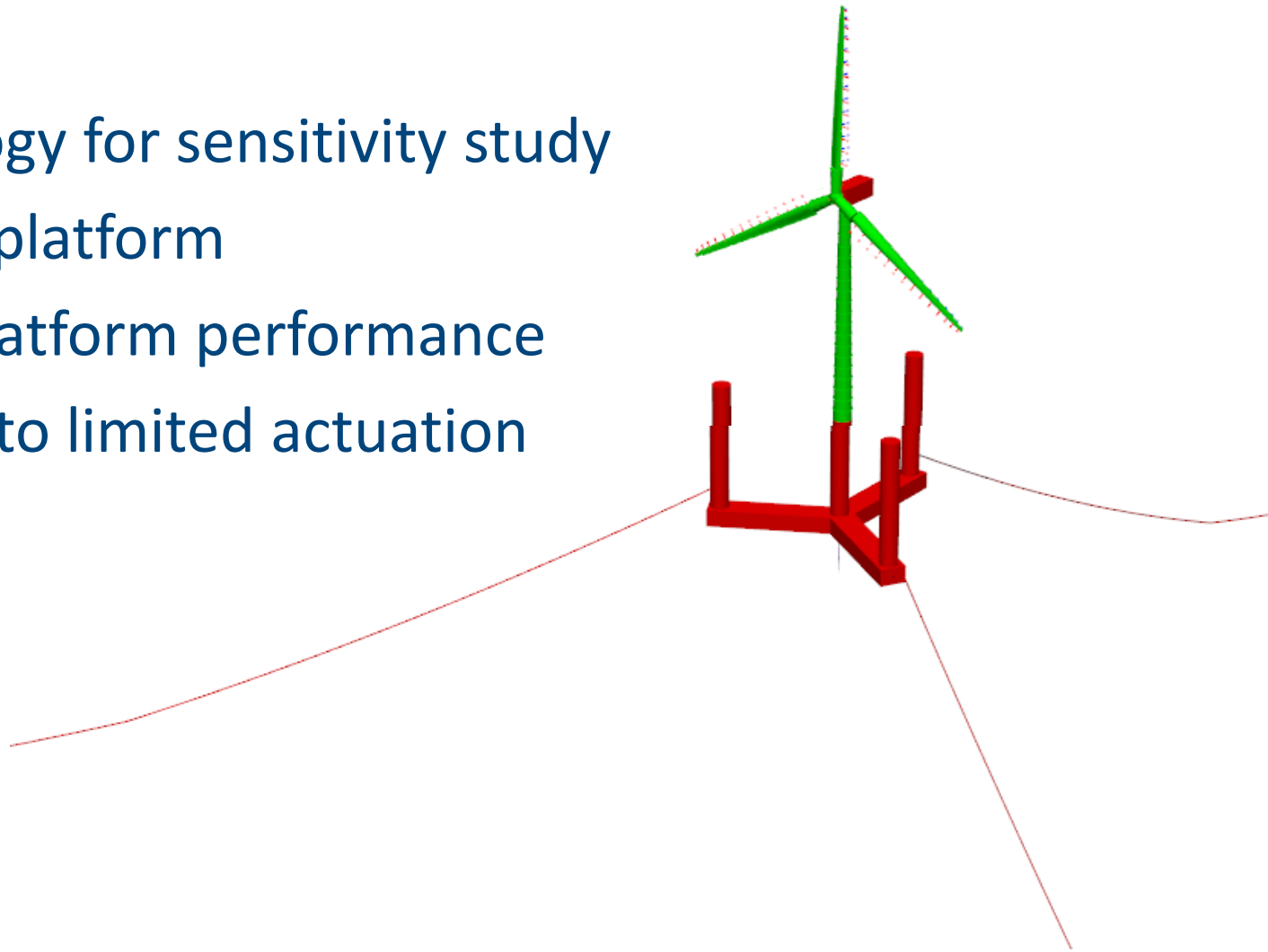
Real-time hybrid model testing of floating wind turbines: **sensitivity to limited actuation**

- What happens if we are not able to actuate all of the forces and moments?

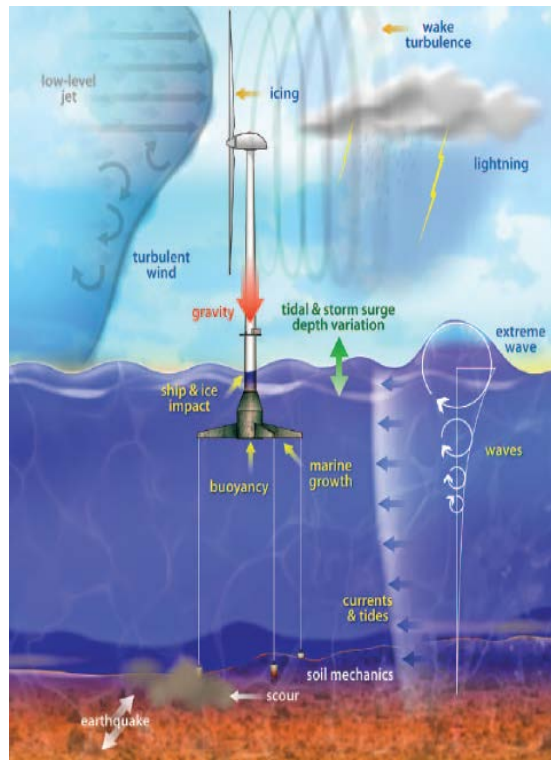


Outline

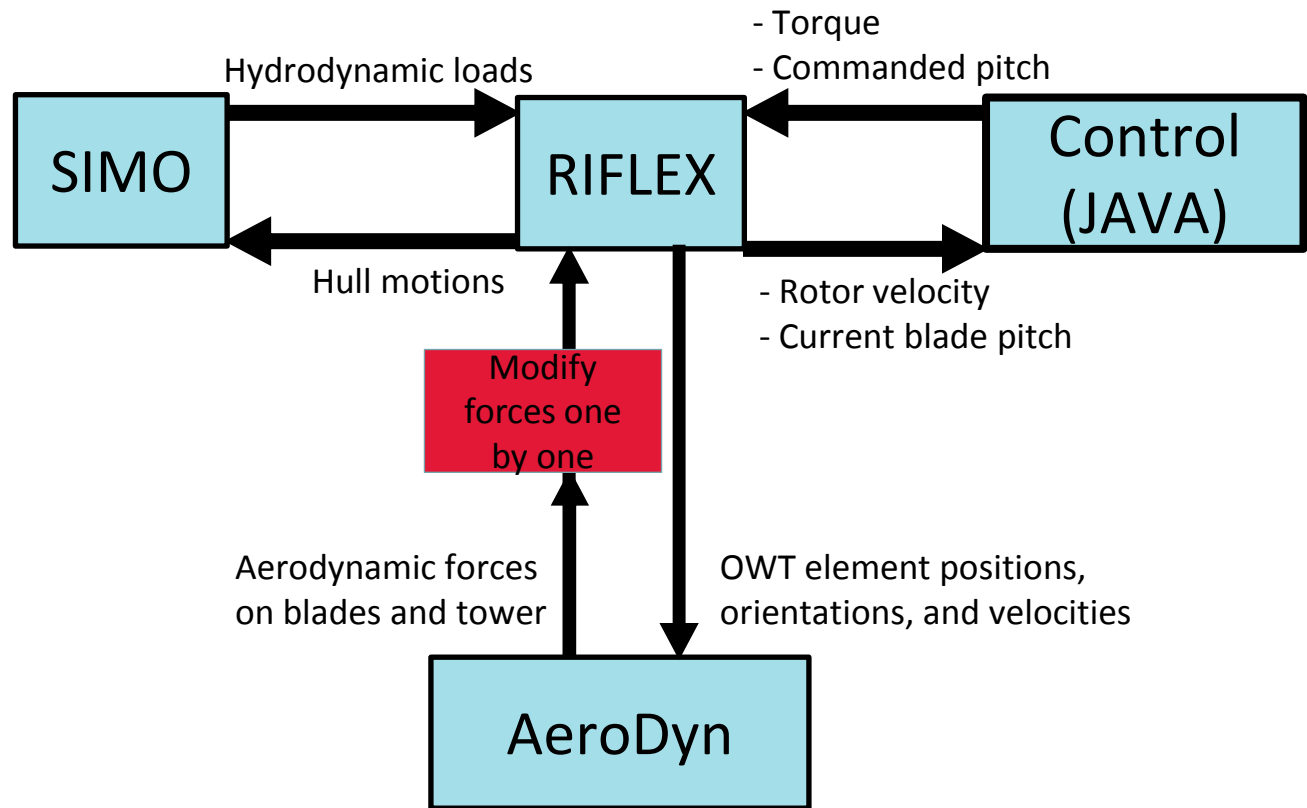
- Methodology for sensitivity study
- 5MW-CSC platform
- Baseline platform performance
- Sensitivity to limited actuation



Computational methodology



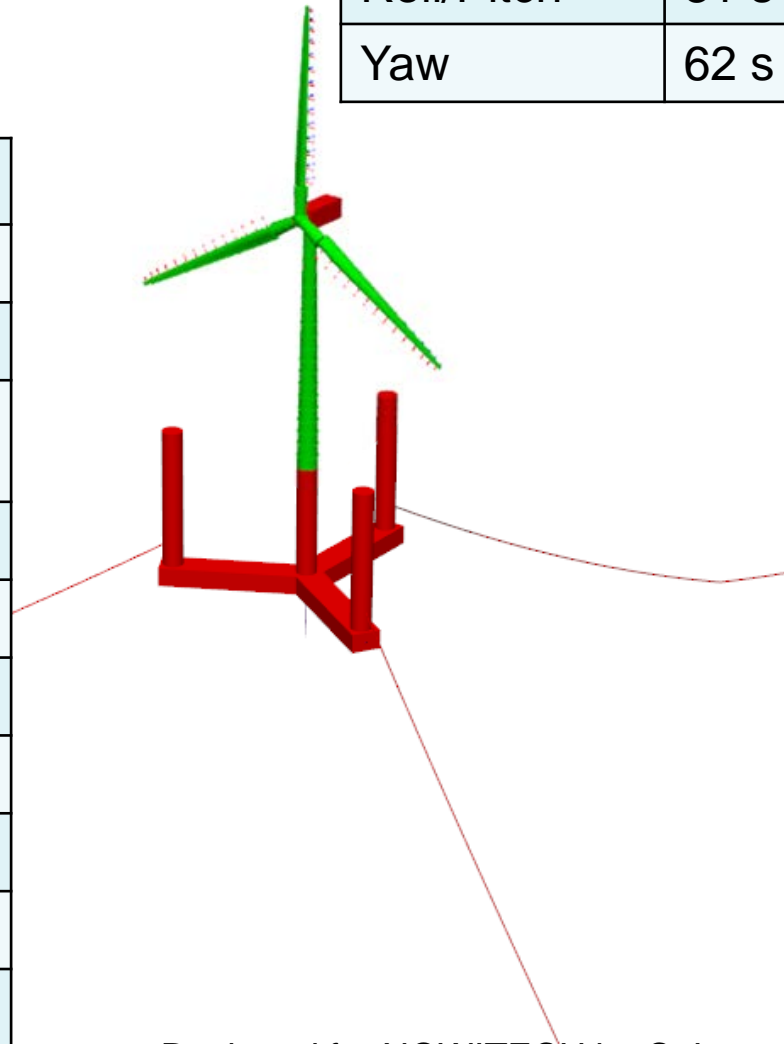
Source: NREL/Wind power today, 2010.



5MW-CSC platform

Surge/Sway	80 s
Heave	26 s
Roll/Pitch	31 s
Yaw	62 s

Diameter of center and offset columns	6.5 m
Pontoon height	6.0 m
Pontoon width	9.0 m
Distance from center column midpoint to pontoon edge	45.5 m
Draft	30.0 m
Freeboard	20.0 m
Water depth	200 m
Anchor point radius	884 m
Mooring line weight (total)	258 tonnes
Hull steel mass	1686 tonnes
Turbine	NREL 5MW

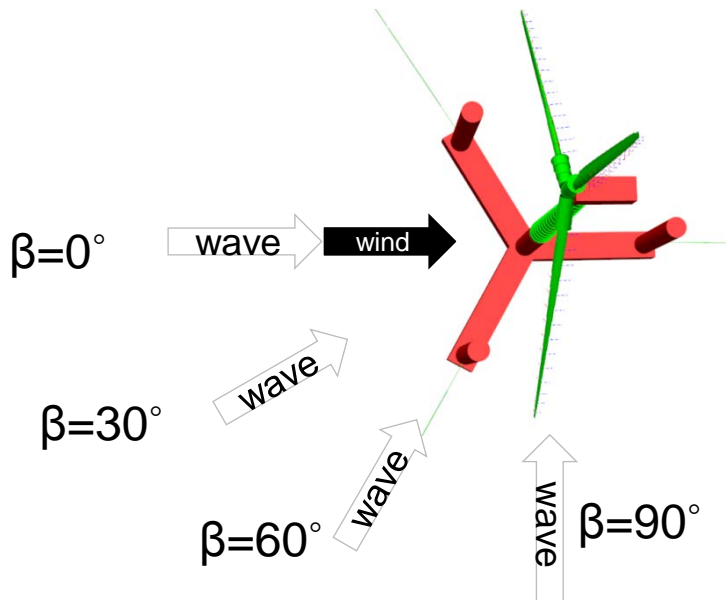


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Environmental Conditions

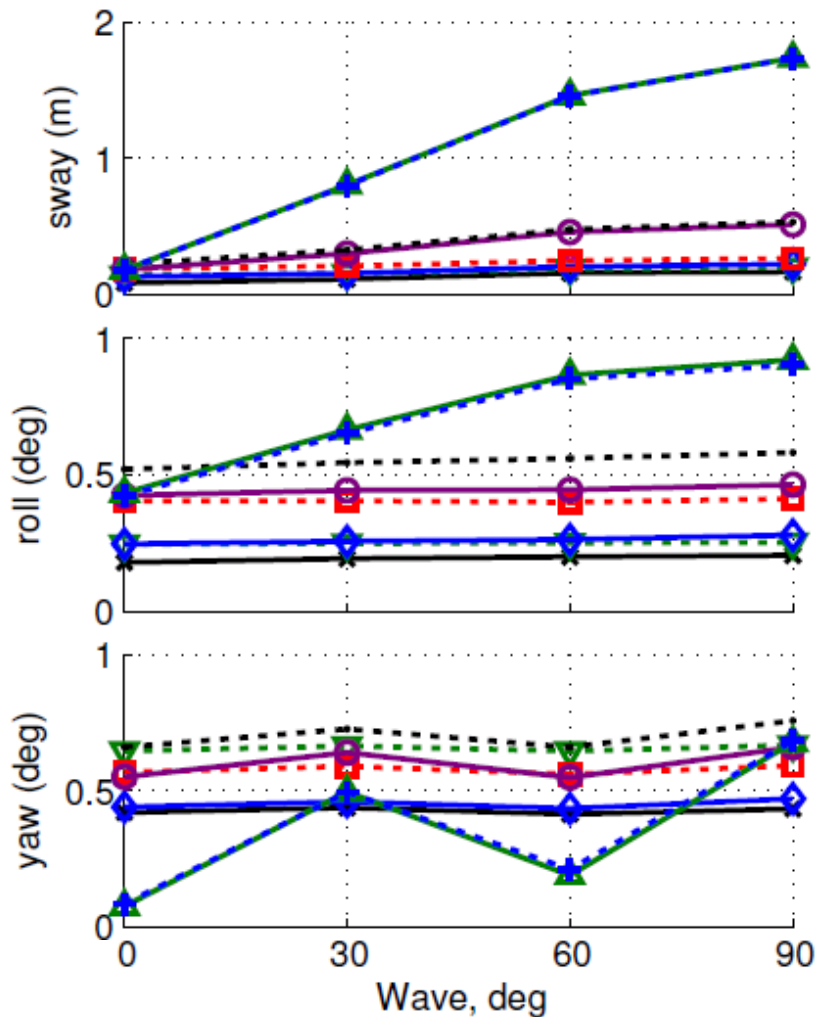
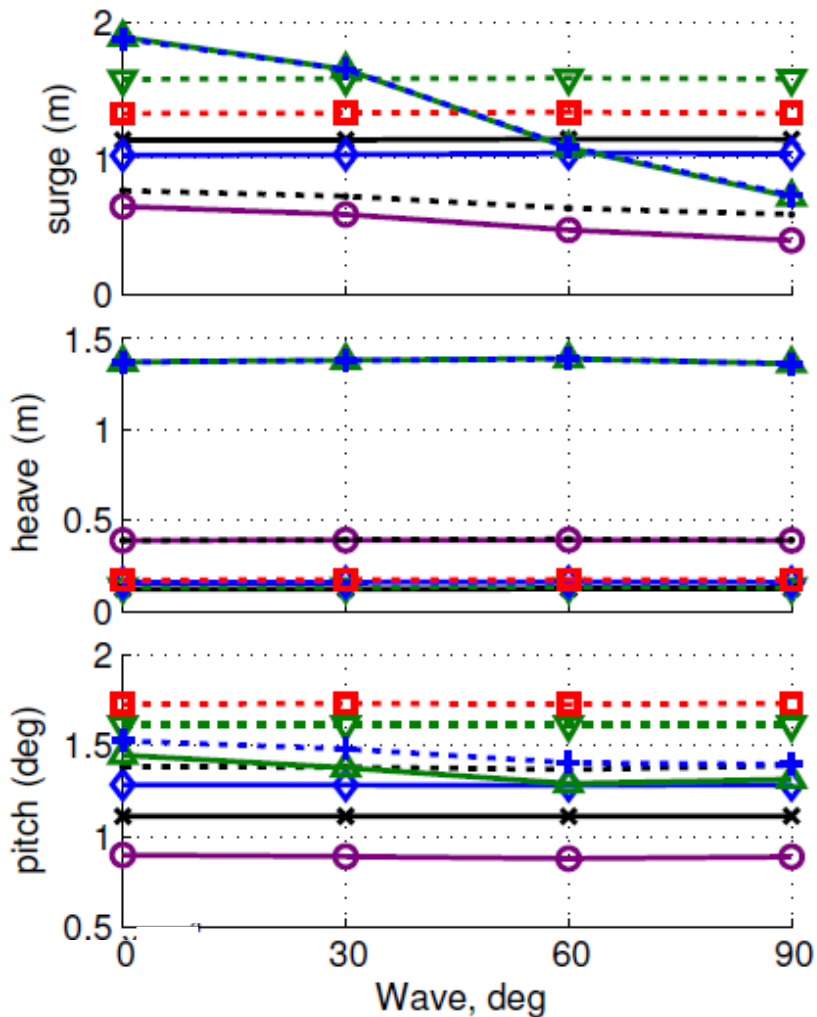
	EC 1	EC 2	EC 3	EC 4
U (m/s)	8.0	11.4	20.0	49.0
H _s (m)	2.5	3.0	5.9	14.4
T _p (s)	9.8	10.1	11.3	13.3
I (%) (NTM)	17.1	14.0	11.5	10.0
I (%) (ETM)	28.1	23.2	15.7	10.7

- Representative below-rated, at rated, above-rated, and storm conditions
- Normal and extreme turbulence considered
- Misalignment between wind and waves
- 1-hour simulations

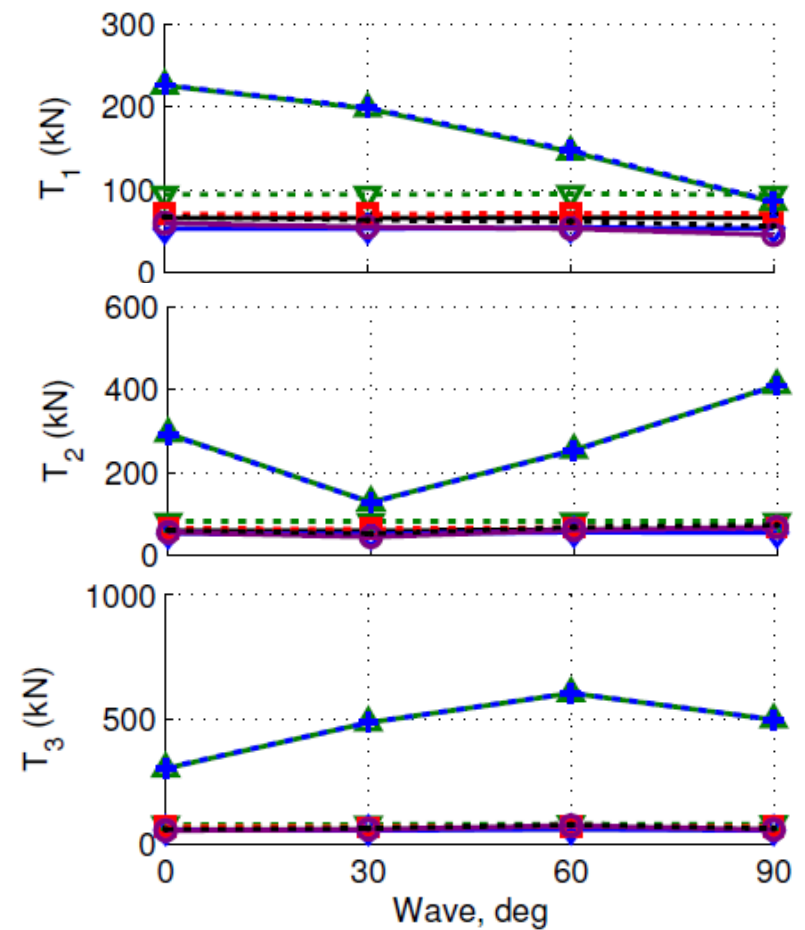
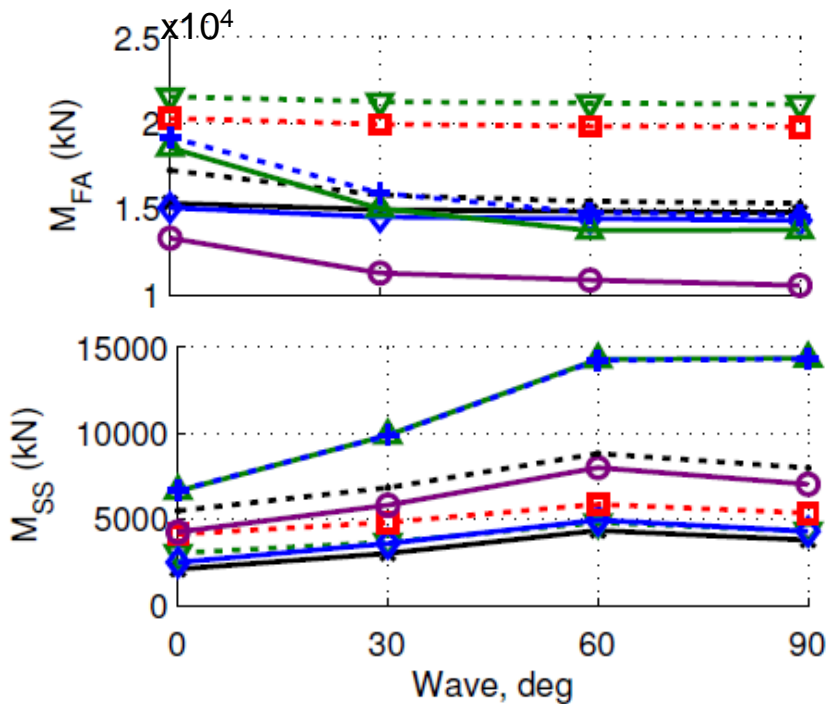


Baseline performance: standard deviation of motions

- EC 1 NTM (solid black line, asterisk marker)
- EC 1 ETM (dashed green line, inverted triangle marker)
- EC 2 NTM (solid blue line, diamond marker)
- EC 2 ETM (dashed red line, square marker)
- EC 3 NTM (solid purple line, circle marker)
- EC 3 ETM (dashed black line, no marker)
- EC 4 NTM (solid green line, triangle marker)
- EC 4 ETM (dashed blue line, plus marker)



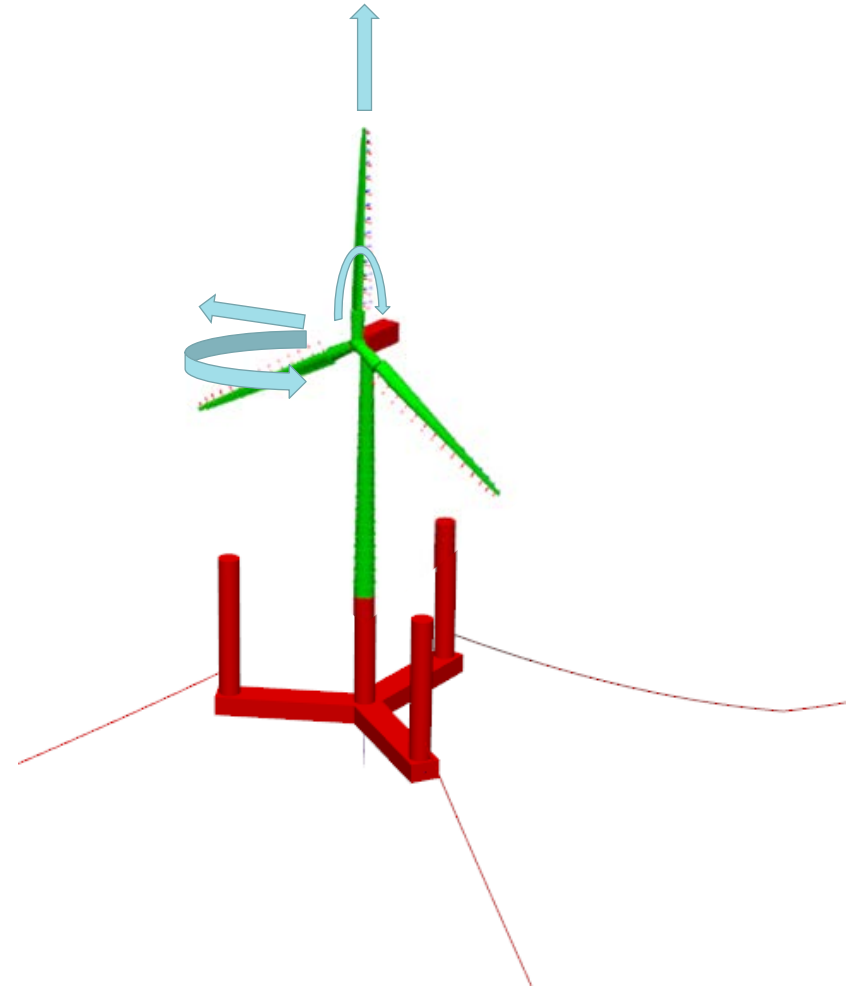
Baseline performance: standard deviation of tower base bending moments and mooring line tension



- EC 1 NTM
- ◇— EC 2 NTM
- EC 3 NTM
- △— EC 4 NTM
- - -▽- - EC 1 ETM
- - -□- - EC 2 ETM
- - -◇- - EC 3 ETM
- - -△- - EC 4 ETM

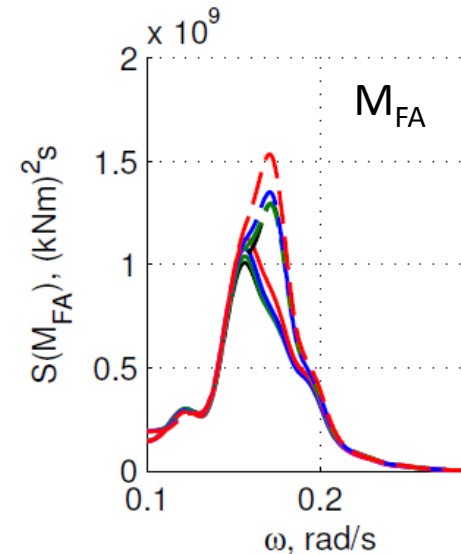
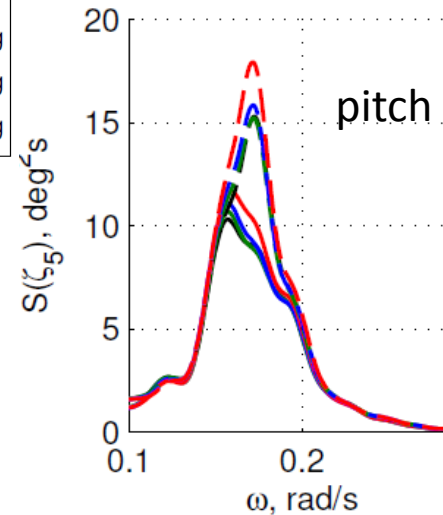
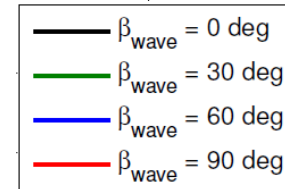
Sensitivity to limited actuation

- non-thrust aerodynamic loads:
 - pitch moment
 - yaw moment
 - sway force
 - heave force
- thrust directionality
- gyroscopic moments
- dynamic variation of generator torque



Results of removing rotor aerodynamic pitch moment

- 1-6% of total aero pitch
- Depends on tilt, wind shear, turbulence, platform motions
- Below rated: affects sway, roll, M_{FA}
- Above rated: affects pitch, line tension and M_{FA}
 - See figure: dashed lines are without aero pitch
 - M_{FA} has a larger wave-frequency component (not shown), which is not affected by aero pitch
 - Effect on tower-bending frequency is also important
 - 10-14 % changes in std dev of line tension and pitch motion



Results of removing aerodynamic sway, heave, and yaw

- Sway:

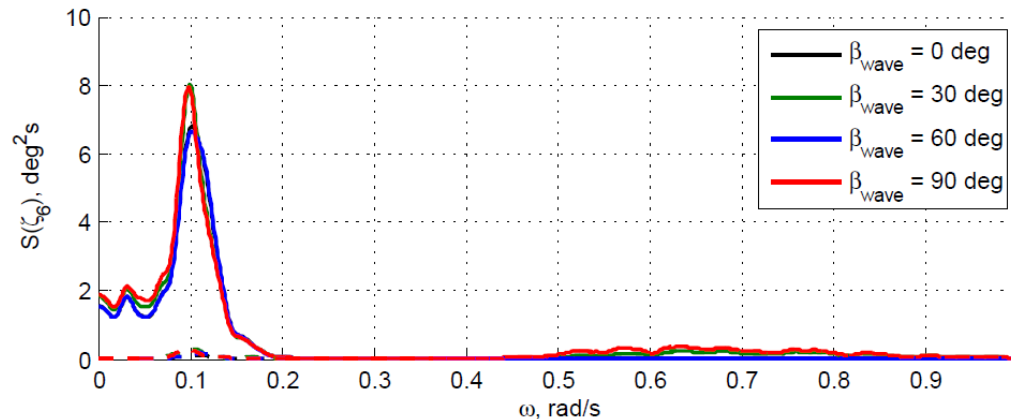
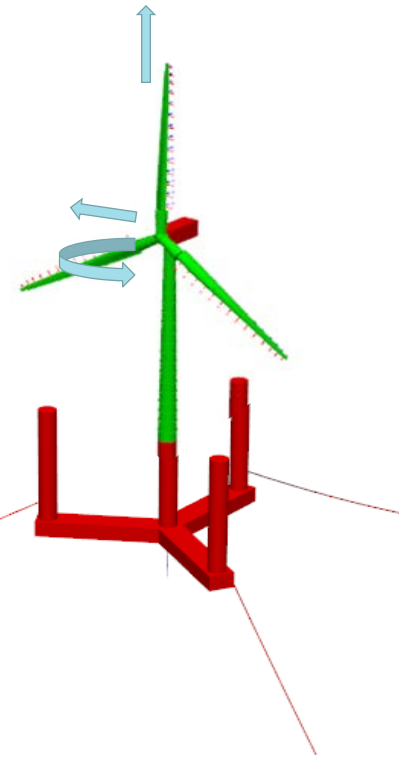
- Sway motion, roll motion, and tower side-side bending moment
- Large relative effect in aligned conditions (due to small absolute values)

- Heave

- <3% effect on key responses in operational conditions
- May be more important in parked (storm conditions)

- Yaw

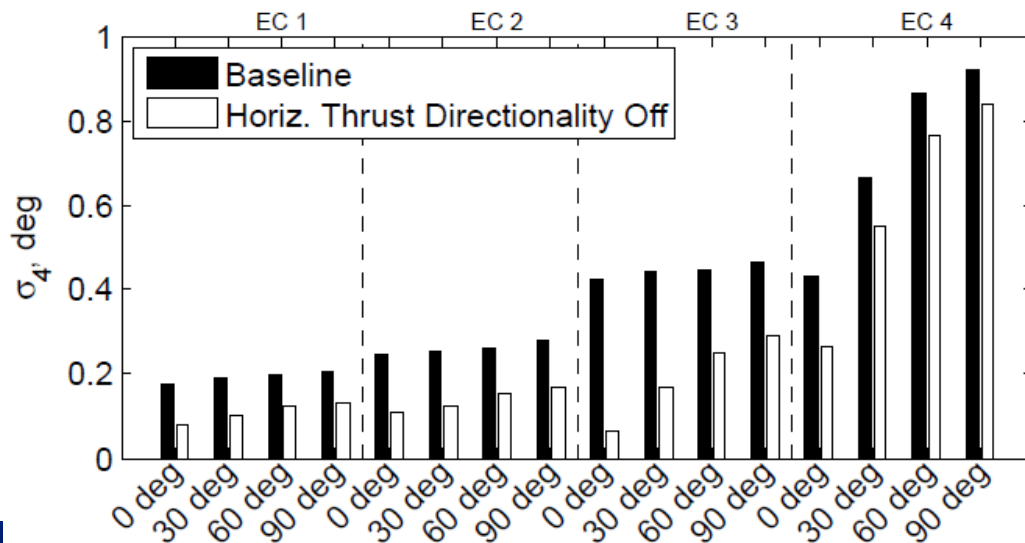
- Large effect on yaw motions at low frequencies (80% change in std. dev)



Results of removing thrust directionality

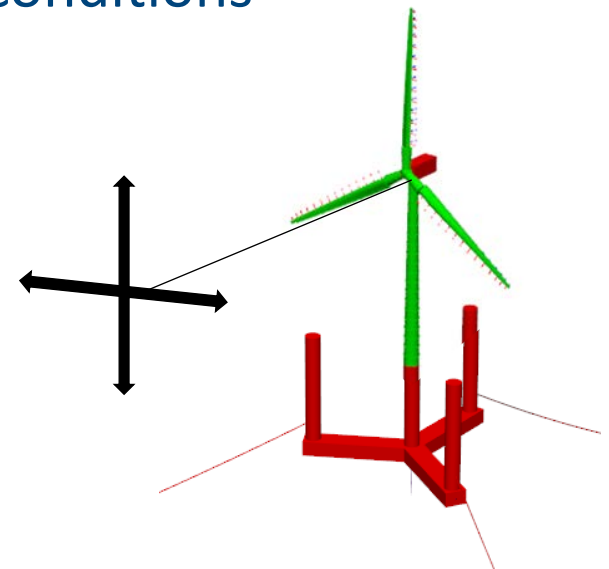
Horizontal directionality

- thrust actuator does not follow the motions of the hub in the horizontal plane
- Affects sway, roll, M_{SS}
- 50% effect on roll std. dev. in cases where roll \approx pitch



Vertical directionality

- Thrust actuator does not follow the vertical hub motions
- Less than 5% effect on key responses in operational conditions



Results of removing gyroscopic moments and dynamic generator torque

- Gyroscopic moments

- Only relevant for the operational conditions
- 5 % change in the std. dev. of the yaw motion
- Effect seen near the yaw and pitch natural frequencies

- Dynamic generator torque

- Control system prescribes constant torque above rated
- Affects roll and M_{SS} in ECs 1 and 2 (4-9% effect on std. dev., but small absolute values)

Summary

- Methodology for investigating the sensitivity of ReaTHM testing of FWTs to limited actuation of aerodynamic forces
- For the given platform, with small motions, non-thrust/non-torque aerodynamic loads had relatively large effects on motions
- Informed trade-offs between complexity and fidelity can be made for experimental design

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